



# IAA 1.9: Atmosphere aerosol remote sensing in the Aerosol-UA project: experimental payload and processing algorithm

Status report for Paris IAA meeting, March 2019

Ya.Yatskiv, O.Degtyaryov, I.Syniavskyi, G.Milinevsky, A.Bovchaliuk,  
M.Sosonkin, M.Mishchenko, V.Danylevsky, Yu.Ivanov, Ye.Oberemok,  
V.Masley, V.Rosenbush, O.Ventskovsky, S.Moskalev, I. Fesyanov

Main Astronomical Observatory, NAS of Ukraine, Taras Shevchenko  
National University of Kyiv, Ukraine, Yuzhnoye State Design Office of  
State Space Agency of Ukraine, NASA Goddard Institute for Space  
Studies, New York, USA, Yuzhnoye Europe Office, Belgium

[genmilinevsky@gmail.com](mailto:genmilinevsky@gmail.com)



# Ukrainian satellite mission Aerosol-UA: atmospheric aerosol polarimetric investigations

Three segments of the project:

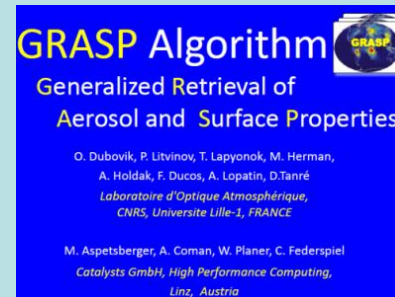
Satellite:  
ScanPol + MSIP



AERONET:  
Validation



Data processing:  
Mission products



Idea for Aerosol-UA project  
come from Glory experiment  
and APS instrument

# Aerosol-UA measurements geometry

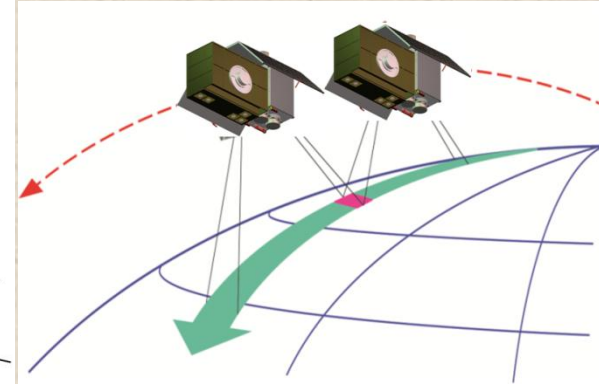
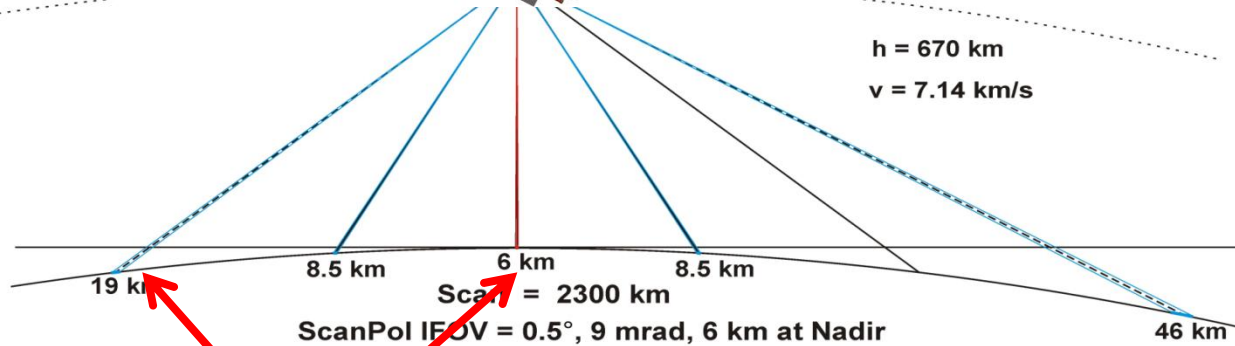
ScanPol



MSIP IFOV = +30 -30 deg (770 km)

ScanPol scan = +50 -60 deg (2300 km)

$h = 670 \text{ km}$   
 $v = 7.14 \text{ km/s}$



ScanPol track

MSIP IFOV = 770km x 770 km

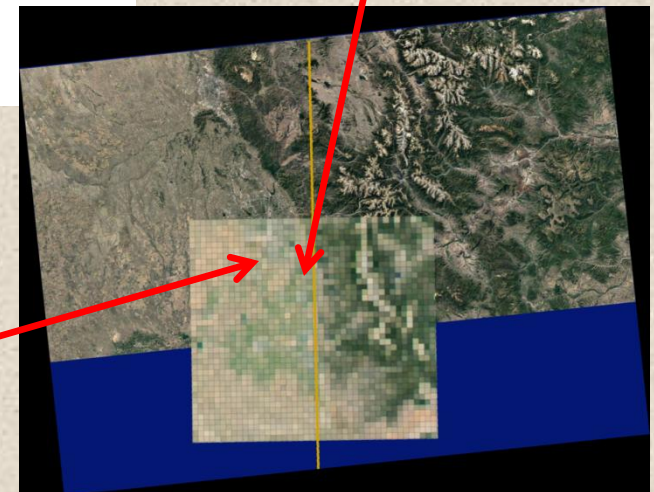
Lengths IFOV along trajectory

19 km 8.5 km 6 km 8.5 km 46 km

MSIP

Field-Of-View  
 MSIP and ScanPol

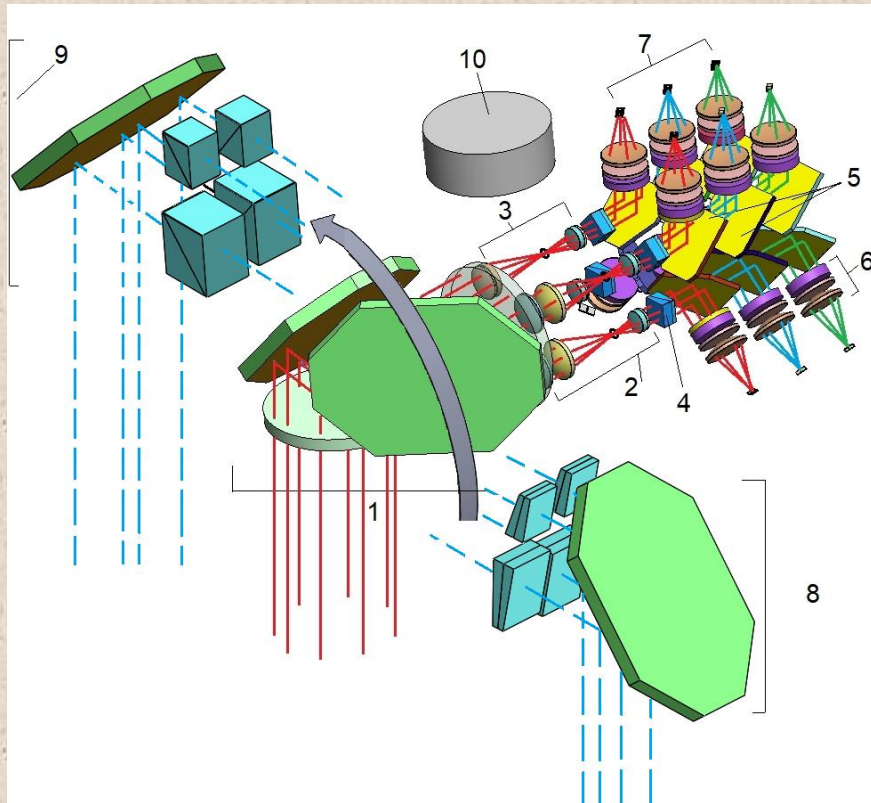
MSIP





# ScanPol polarimeter final optical design, updated in 2019

Spectral band: 370-1610 nm,  
six spectral channels:



**370 nm** - tropospheric  
aerosol and top of clouds

**410 nm** - aerosol over ocean  
and surface

**555 nm** - aerosol over ocean  
and surface, ocean color

**865 nm** - aerosol over ocean  
and surface

**1378 nm** - separate cirrus  
clouds, stratosphere aerosol,  
separation of troposphere and  
stratosphere aerosol in case  
of volcanic eruption

**1610 nm** - separation  
surface signal from aerosol  
over Earth' surface

Filter  $\frac{1}{2}$  width 20 - 60 nm

Observable Stokes parameters: I, Q, U (**0,90,45,135°**)

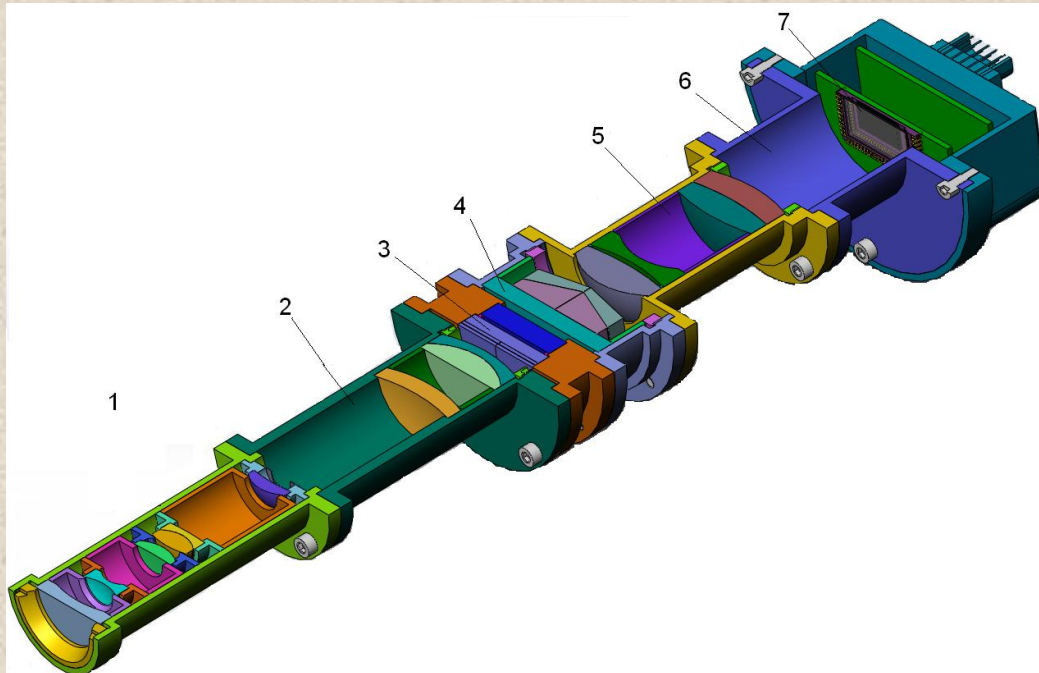
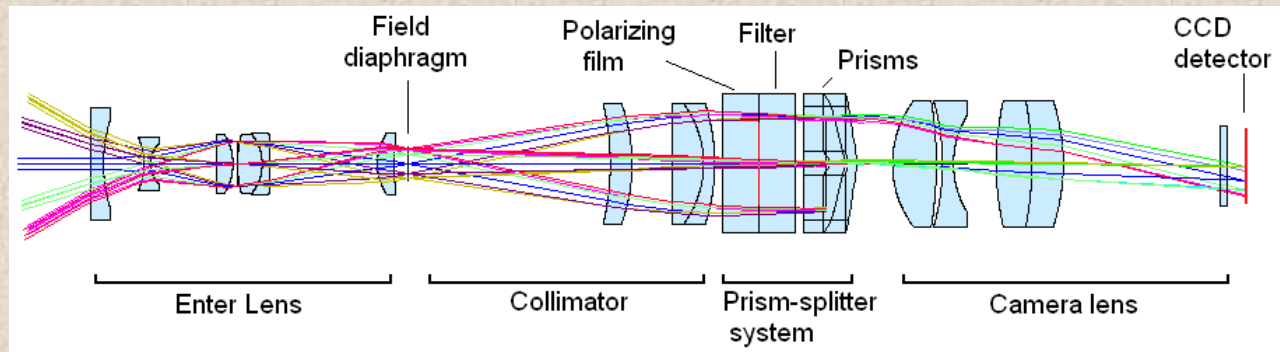
Photometric accuracy: 4%

Polarimetric accuracy: **0.15%**

On-board calibration: all three Stokes parameters

**ScanPol is similar to  
APS Glory**

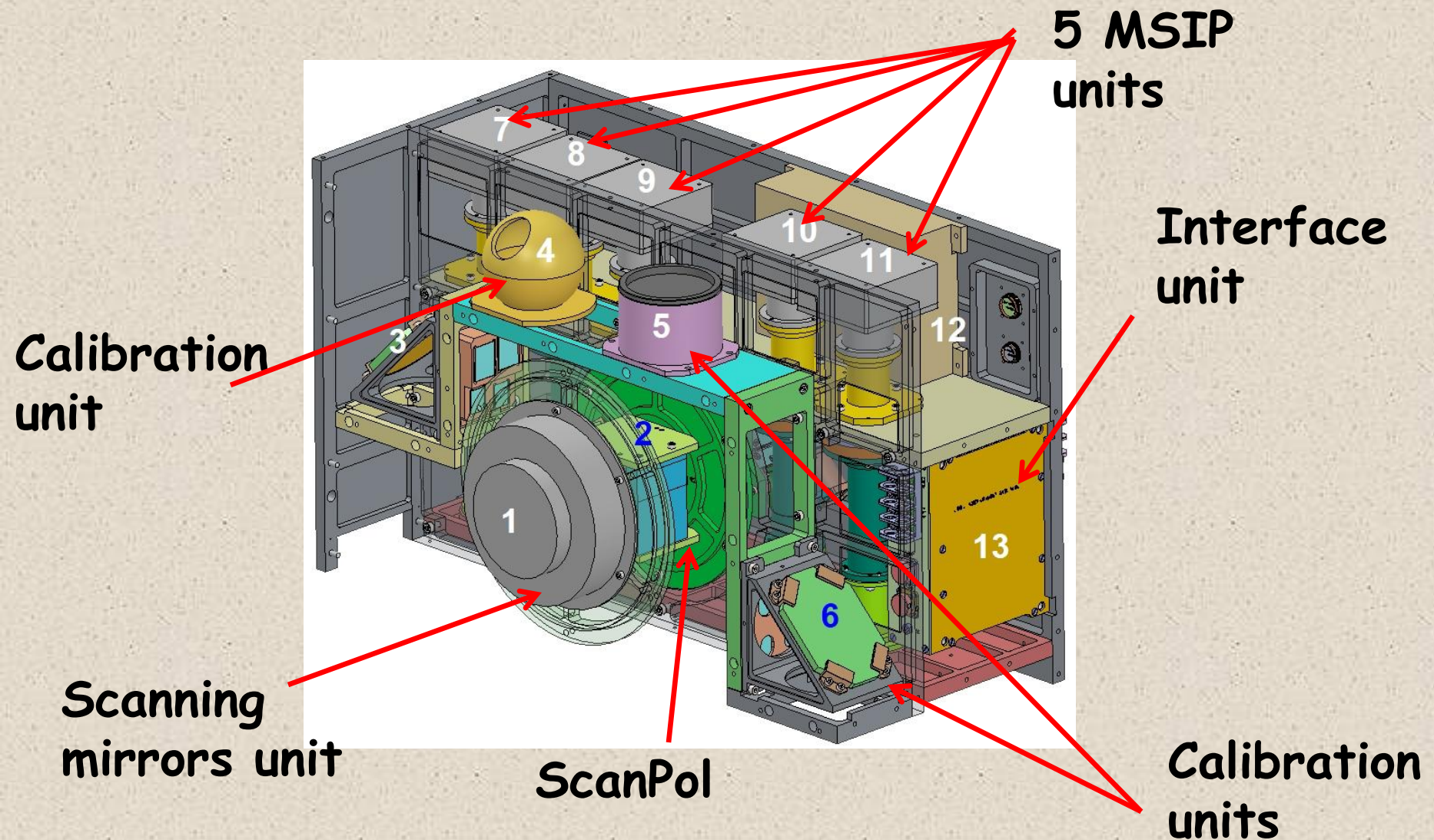
# MSIP optic modeling



- 1 - input lens;
- 2 - collimator;
- 3 - polarizer and filter;
- 4 - prism image splitting system;
- 5 - camera lens;
- 6 - junction;
- 7 - CMOS sensor



# ScanPol and MSIP polarimeters: corrected final design for YuzhSat platform



# ScanPol data processing structure

## Level 0 (raw data)

### ADC output:

6 channels per  
4 Intensity parameters  
= 24 parameters  
+ time

### Calibration parameters:

Depolarizer ~5  
Polarizer 1 parameter  
Black body ~5  
Diffuser plate 1 param

### Telemetry:

1) temperatures,  
voltages and states  
2) data timing,  
spacecraft altitude and  
ephemeris data

## Level 1B

### Calibrated:

24 parameters  
+ time

### Telemetry:

data timing,  
spacecraft altitude and  
ephemeris data

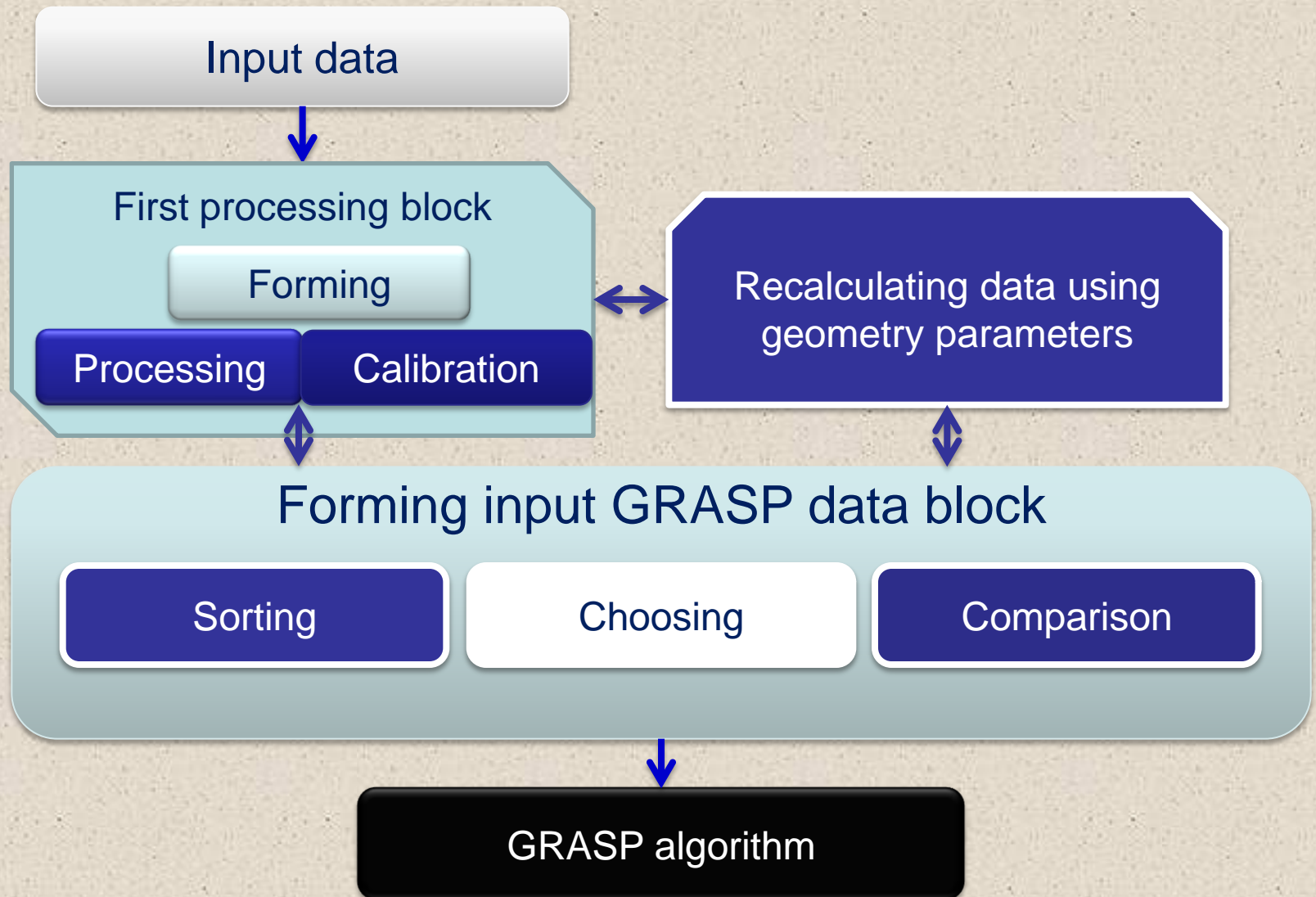
## Level 1C

I, Q, U for  
6 channels  
+ time  
+solar zenith angles  
+zenith obs. angles  
+azimuth obs. angles  
+cloudy  
+x\_coord\_longitude  
+y\_coord\_latitude  
+MASL  
+land percent  
+gas information  
....



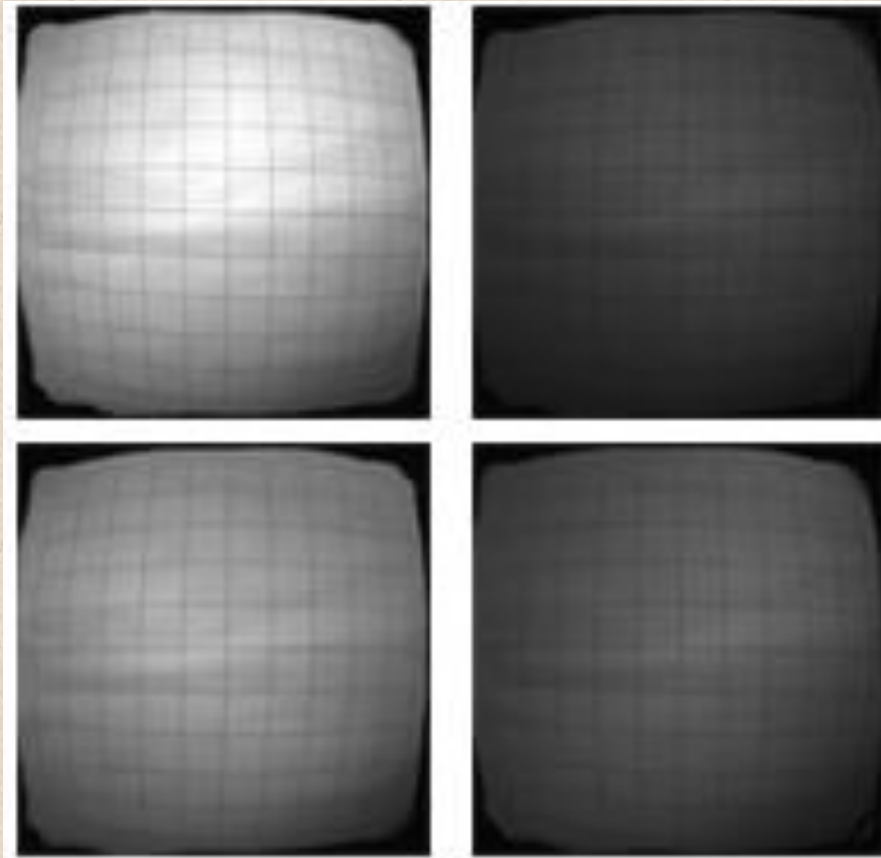
GRASP

# MSIP data processing scheme

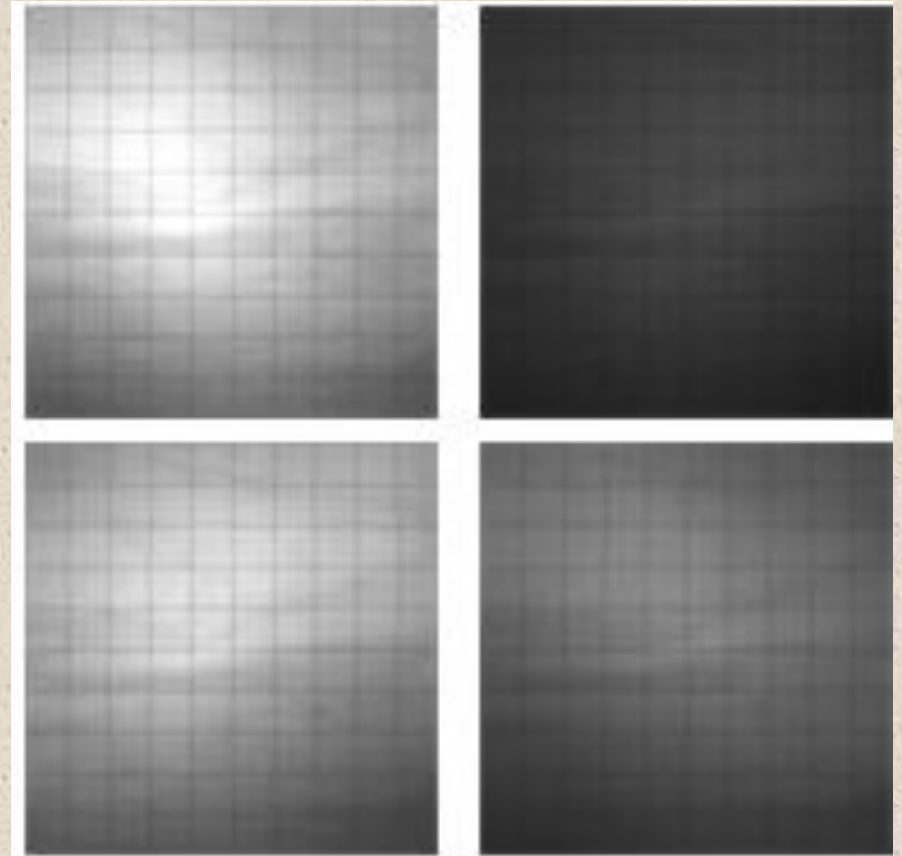




# MSIP image distortion compensation

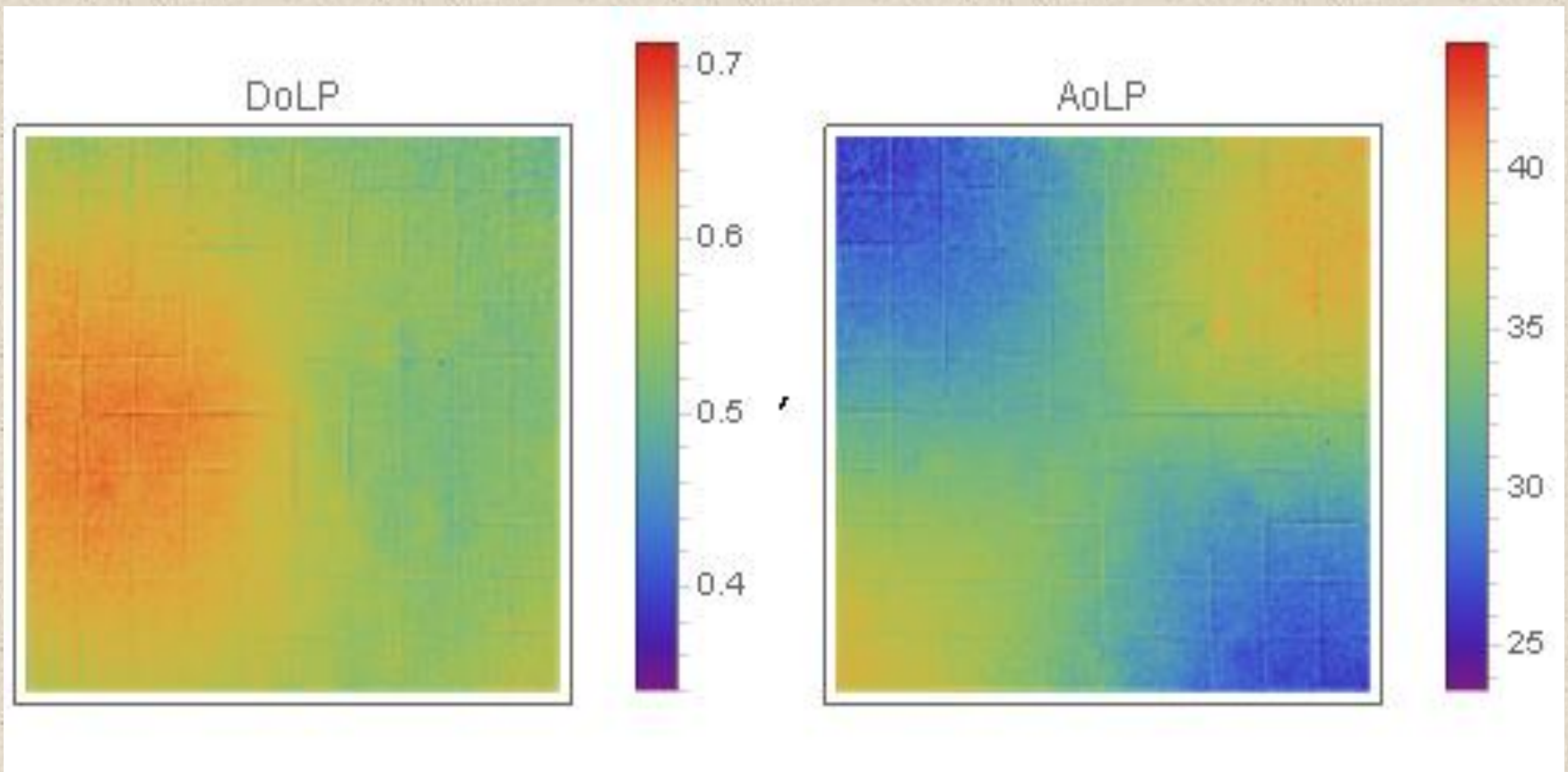


Use of gauge grids (view in polarized light)



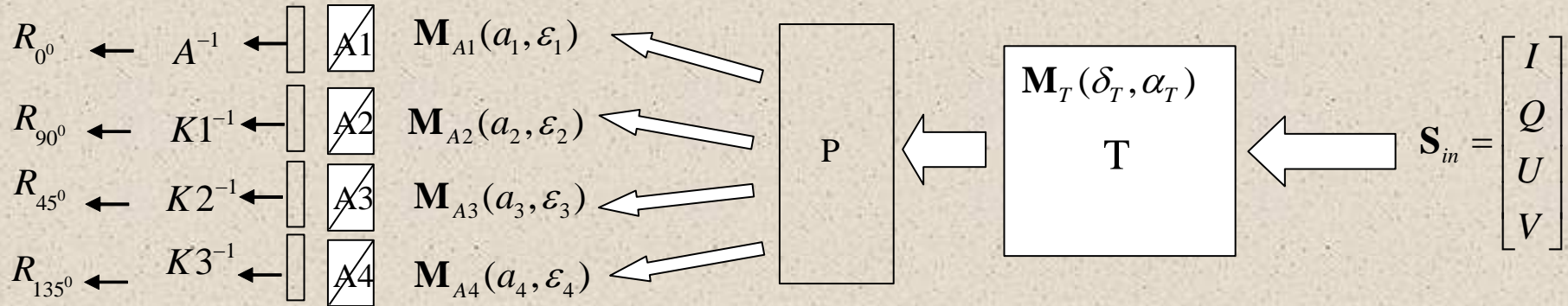
After applying the distortion compensation algorithm

# Polarization calibration of MSIP



**Determination of degree (DoLP) and angle (AoLP) polarization of linearly polarized input light in an uncalibrated MSIP channel**

# MSIP data processing model



Muller matrix method is used.

The model takes into account:

polarization distortion of lenses and film polarizers

the difference of the polarization-independent light intensity transfer coefficients in the conjugate polarization sub channels (A, K)

Distortion compensation algorithm



# MSIP polarization light calibration

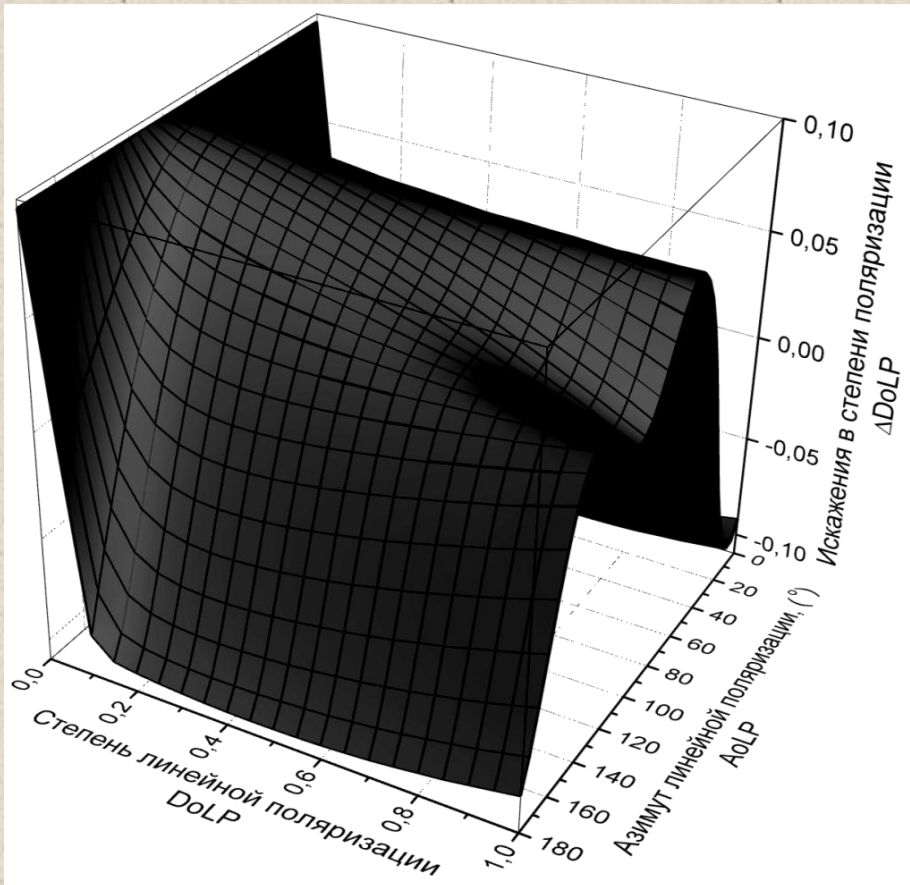
**General analytic expression for a signal detected in a pixel**

$$\begin{aligned} RD_{\alpha_A} &= K^{-1} \times \left[ M_A(a, \varepsilon) \times M_T(\delta, \alpha) \times \begin{bmatrix} I & Q & U & 0 \end{bmatrix}^T \right]_0 = \\ &= \left[ Q \left[ \cos(2\alpha_A + \varepsilon) \left[ (\cos(\delta) - 1) \sin^2(2\alpha) + 1 \right] - \frac{1}{2} \sin(2\alpha_A + \varepsilon) \sin(4a) (\cos(\delta) - 1) \right] + \right. \\ &+ U \left[ \sin(2\alpha_A + \varepsilon) \left[ (\cos(\delta) - 1) \cos^2(2\alpha) + 1 \right] - \frac{1}{2} \cos(2\alpha_A + \varepsilon) \sin(4a) (\cos(\delta) - 1) \right] \left. \right] K^{-1} a^{-1} + \\ &+ IK^{-1} \end{aligned}$$

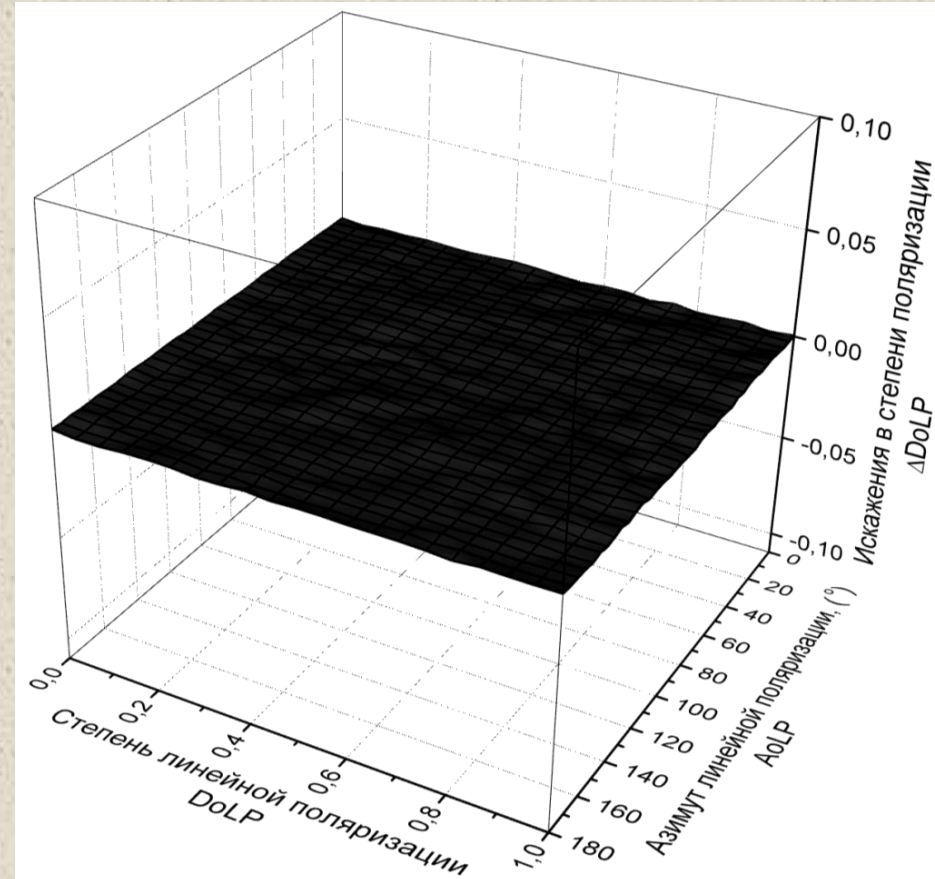
**To estimate the calibration parameters and compensate for polarization distortion, it is planned to use reference sources with linearly polarized light.**

# MultiSpectral Imaging Polarimeter test measurements

DoLP corrections



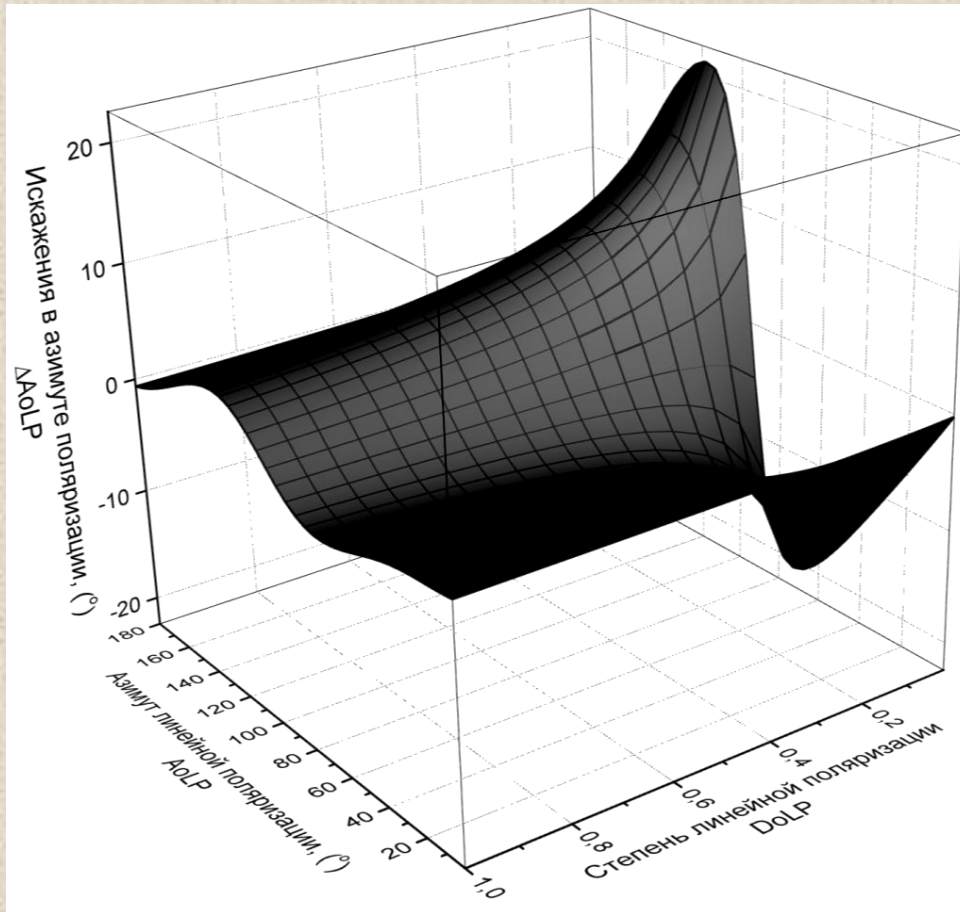
Before calibration



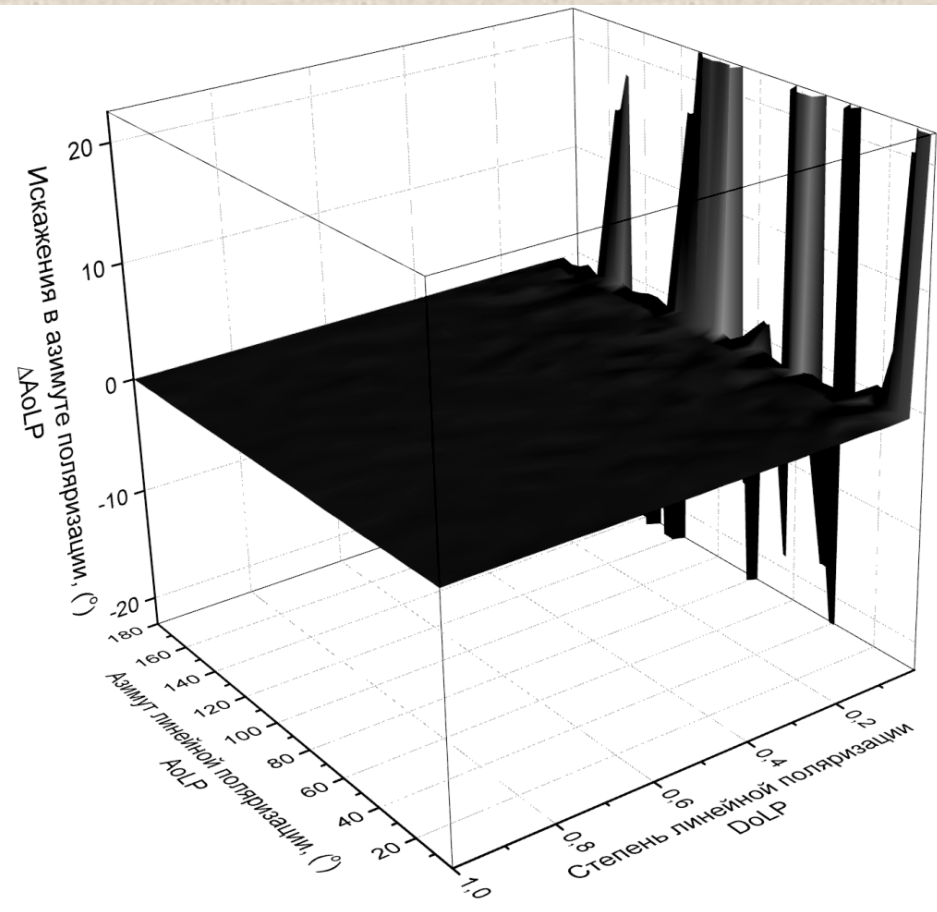
After calibration

# MultiSpectral Imaging Polarimeter model calibration

AoLP



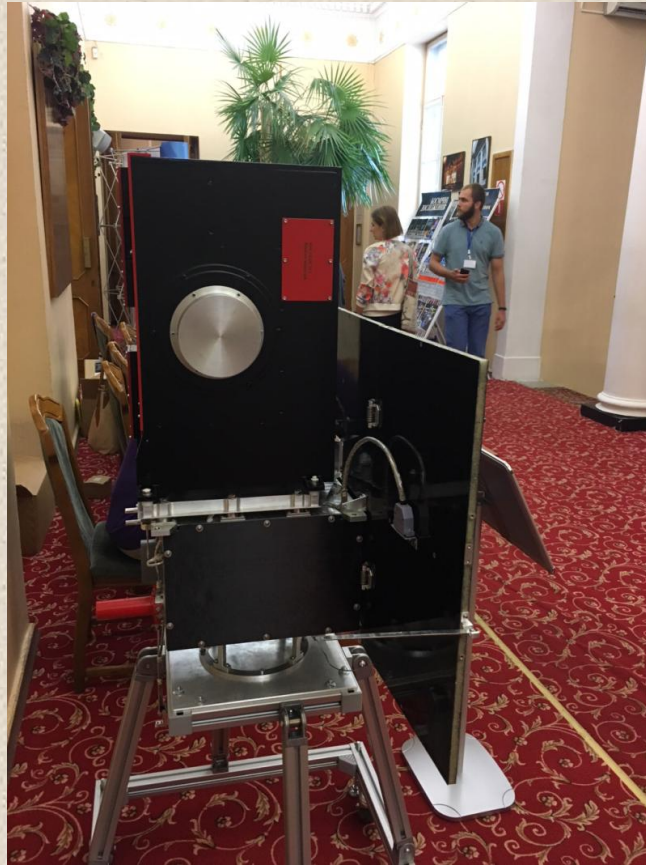
Before calibration



After calibration



# ScanPol and MSIP polarimeters onboard of YuzhSat platform



Satellite platform YuzhSat  
designed by State Design  
Office "Yuzhnoe"

## Characteristics of payload

### Orbit

Type: sun-synchronous

Inclination:  $\sim 98^\circ$

Altitude:  $\sim 705$  km

### YuzhSat platform:

Pointing accuracy:  $\sim 0.1^\circ$

Total mass of scientific payload  
estimated:  $\sim 22$  kg

Power for payload:  $\leq 25$  W

Design life:  $> 3$  years

# Updated Timeline

**Aerosol-UA mission has been included into State Space Program - 2018**

**State Space Program to be adopted - 2019**

**Aerosol-UA payload experimental model - 2019**

**Aerosol-UA flight payload ready - 2020**

**Aerosol-UA flight payload testing - 2021**

**Aerosol-UA launch (planned) - 2022**