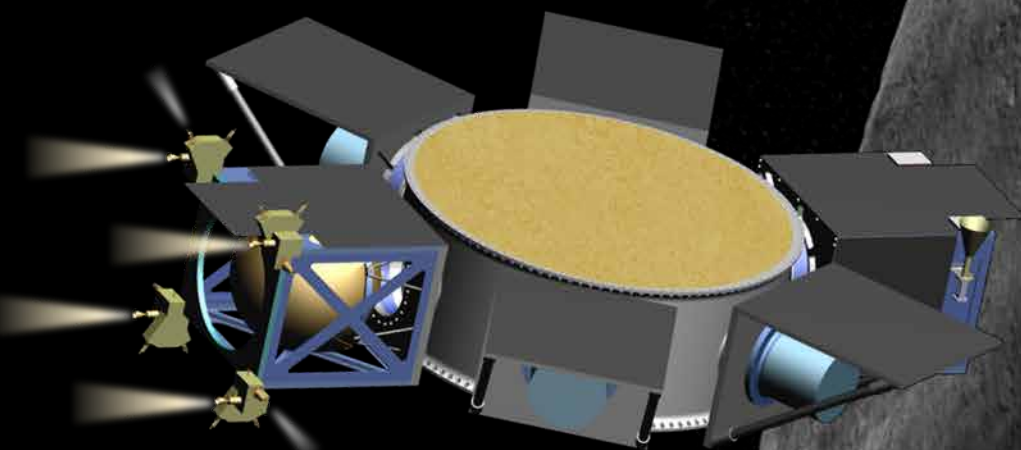


ISIS

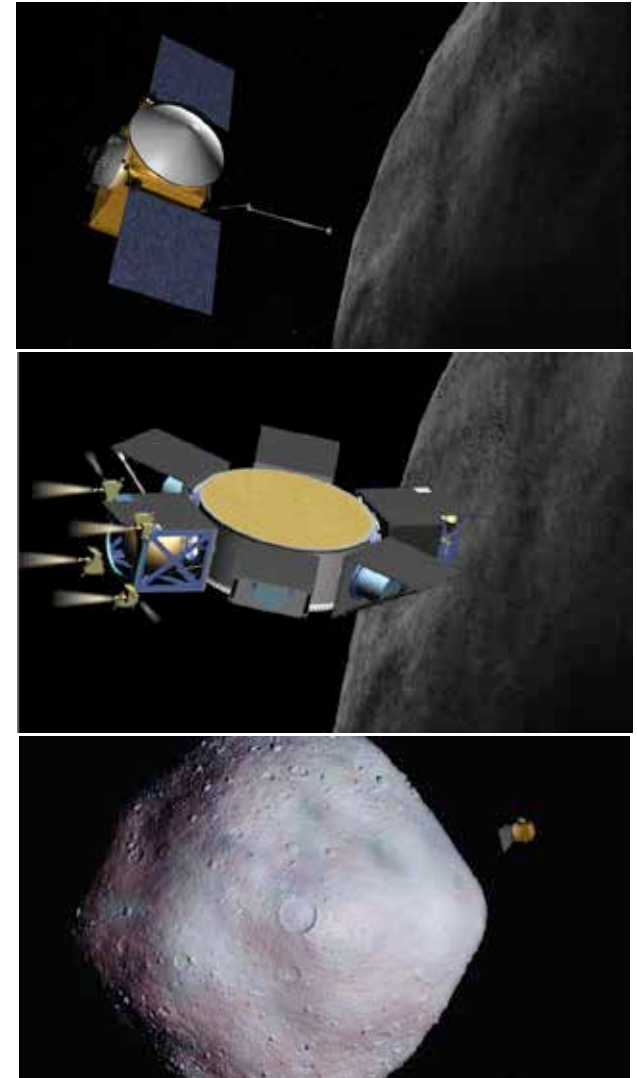
Impactor for Surface and Interior Science



ISIS Mission Concept



- } Send an independent, autonomous impactor spacecraft, to the asteroid target of the OSIRIS-REx mission
 - } Co-manifest impactor with InSight launch
 - } Arrive after OSIRIS-REx has completed its science objectives (i.e., sample collection)
- } **ISIS creates crater tens of meters in diameter**
 - } OSIRIS-REx images the impact from a safe vantage point (~1-meter resolution)
- } **Seismic reverberations throughout the asteroid cause global modifications**
 - } After debris clears, approach asteroid for imagery of crater and previously mapped terrain (~2 cm resolution). Also collect spectra of pristine material exposed by impact
- } **Deflection experiment**
 - ▶ ²} Measure asteroid delta-V due to impactor



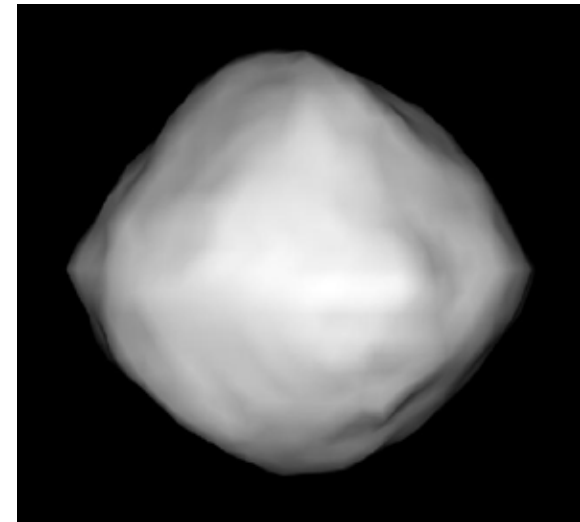
Cross-cutting Exploration & Science Benefits



ISIS delivers Discovery-level Science, closes Exploration SKGs and demonstrates NEO Mitigation Technology, all for a small fraction of the cost of a

Discovery mission

- } Cratering experiment – regolith properties and geophysics
- } Seismic experiment – global alterations (toppled rocks, landslides) due to shock wave/reverberations, lofting of material far from impact site
- } Ejecta – size distribution of regolith, understanding meteorite formation process
- } Imaging of crater – formation processes, morphology and subsurface geology
- } Spectroscopy of pristine material from depth provides context to OREx sample
- } Characterization of any volatiles released from impact site
- } Thermal properties of disturbed & undisturbed areas
- } Topographic mapping before and after to reveal exhumed volume and material mobility
- } Particulate environment following impact over a wide range of disturbance energies
- } Rotational excitation can constrain the interior mass distribution.
- } Measure delta-V imparted by impact – determine momentum enhancement due to ejecta

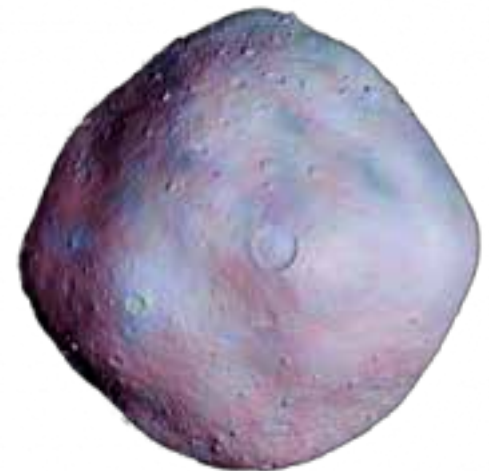


The Planetary Defense aspects of a deflection experiment will generate significant public interest.

ISIS/ORIRIS-REx Operations Concept

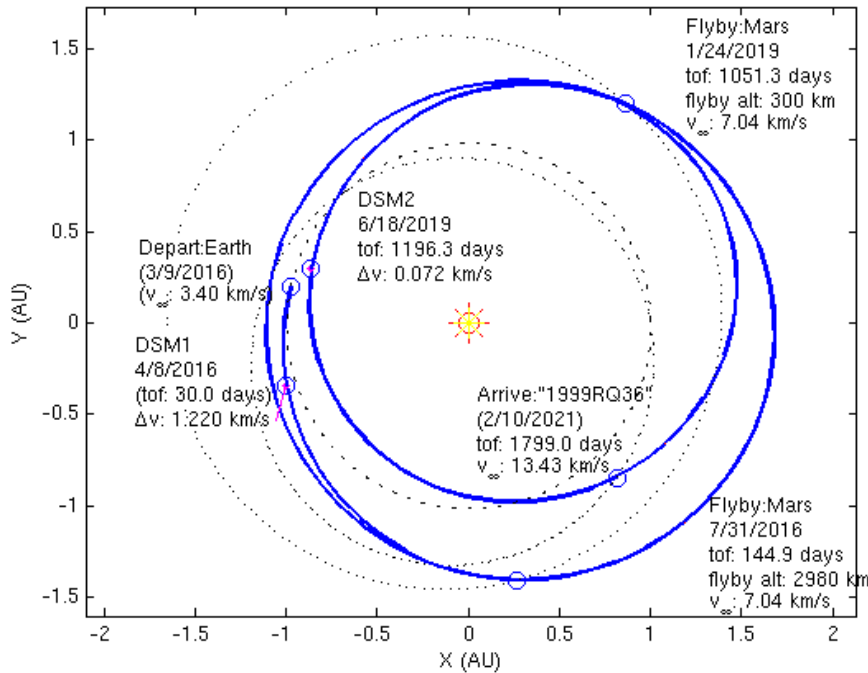


- } Pre-impact characterization of asteroid ephemeris
- } Move to safe observing location and image ejecta cone as it expands over a period of several minutes.
 - } 50 km gives 0.7 m/pixel with Polycam
- } Monitor ejecta as it dissipates (15-20 days)
- } Perform slow flyby(s) for imaging and spectra (15-20 days)
- } Enter radio science mode
 - } Few km terminator orbit for 15-20 days
- } Total time from impact 45-60 days
 - } Assume **90 days** science operations from impact to departure, including margin

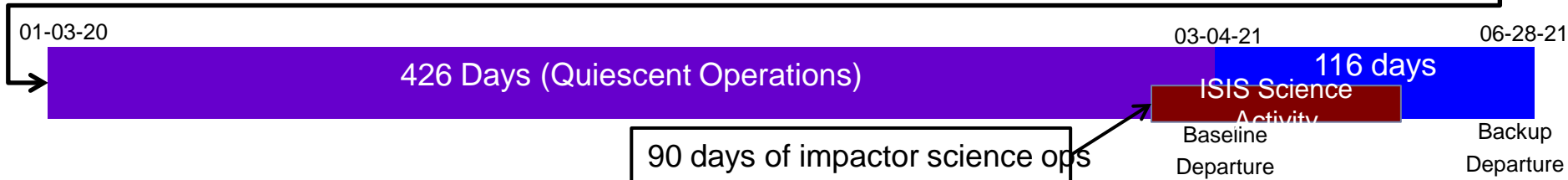
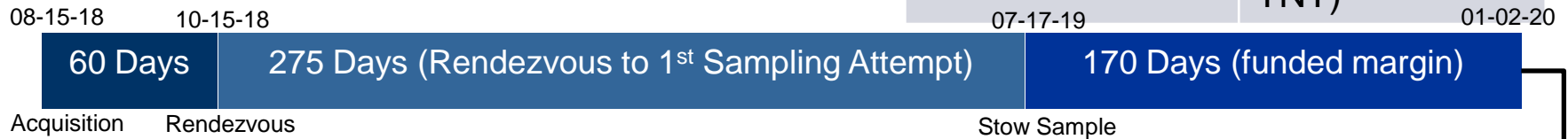




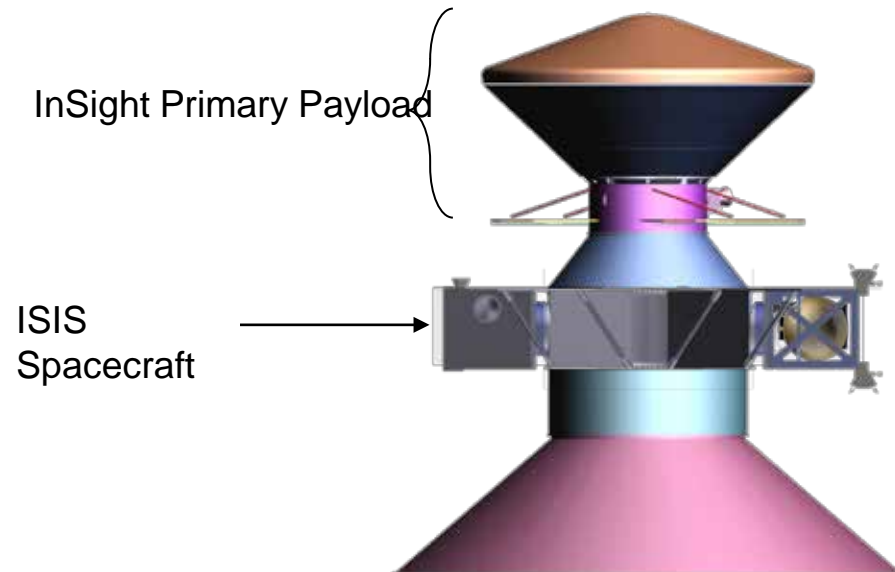
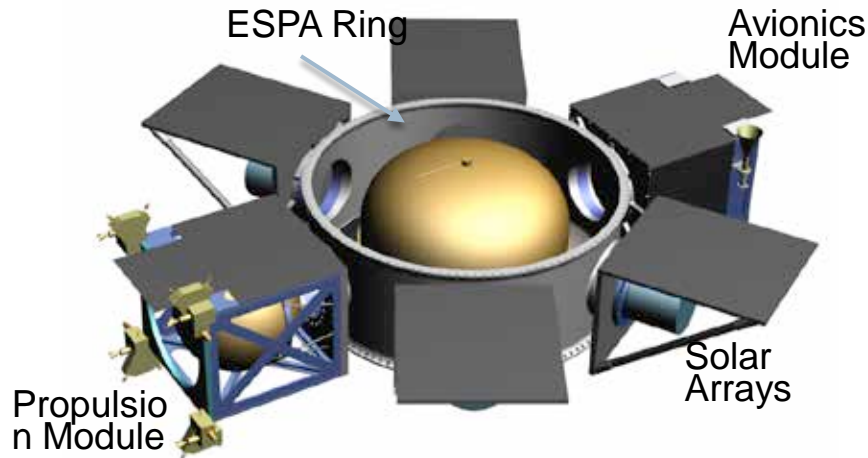
ISIS Schedule Compatible with OSIRIS-REX



ISIS Baseline Mission	
Launch	9-Mar-2016
Mission ΔV	1.29 km/s
Arrival & Impact	10-Feb-2021
Arrival Phase Angle	9°
Impact Velocity	13.4 km/s
Impact Mass	440 kg
Impact Energy	50 GJ (~9t TNT)



ISIS Flight System Overview



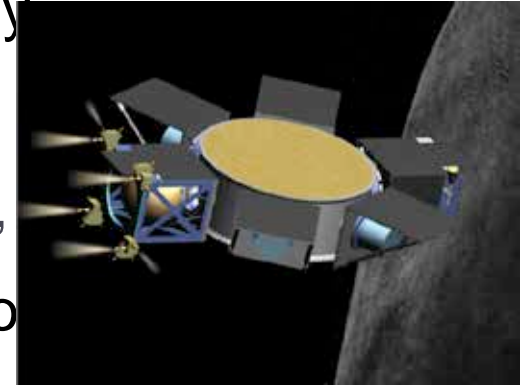
- } System designed around flight-qualified ESPA
- } Imposes no impact on host SC/LV interface
- } Modular Flight System
- } Spacecraft Architecture emphasizes simplicity and reliability
- } No Comm. Crosslink (to observer s/c)
- } No Pyrotechnics
- } No Deployments
- } No Mechanisms





Potential Stakeholders

- } Science – This mission scheme is extraordinarily cost effective and aligned with NASA Science Mission Directorate objectives
 - } Planetary Decadal Report, “The first and most important [criterion] was science return per dollar.”
- } Technology – NEO Mitigation is a Space Technology Grand Challenge
 - } Terminal guidance demonstration
 - } Earth impactor deflection demonstration
- } Exploration
 - } ISIS squarely addresses numerous “Critical” Strategic Knowledge Gaps (SKGs) for human exploration of NEAs
 - } Characterize geotechnical properties, particulate environment, mechanical response, structural integrity and local stability of sub-km NEAs



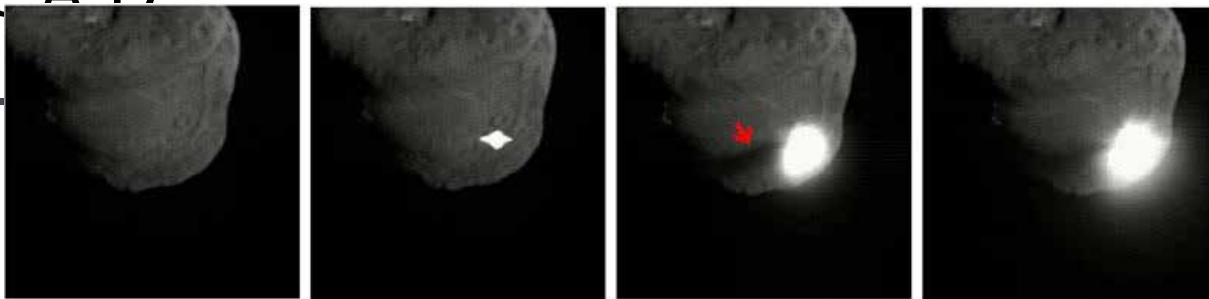
ISIS is a natural candidate for joint funding. In the current climate, this may be imperative in order to take advantage of the InSight secondary payload opportunity.



Schedule

- } Time is short!
 - } The InSight launch provides a critical constraint on the viability of the ISIS mission concept
- } Starting 3-month pre-phase A study
 - } High concept maturity crucial for short development schedule
 - } Leading to a mid-2013 Decision Point
- } ISIS development schedule assumes ~30 month Phase A-D

} Aug.



ht launch

Conclusions



- } ISIS is a low-cost mission that addresses NASA strategic goals and provides Discovery-class science returns across a wide range of small body science disciplines.
 - } The mission leverages NASA's investment in the OSIRIS-REx mission and takes full advantage of the New Frontiers-class instrumentation on the observer spacecraft.
 - } Co-manifesting with InSight further improves the cost-effectiveness.
- } NEAR-Shoemaker is NASA's only NEA rendezvous mission so far. The second will be OSIRIS-REx, twenty years later.
 - } The convergence of OSIRIS-REx schedule and InSight launch opportunity is an extraordinary alignment that will not be repeated again soon.

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- ▶ 9} ISIS represents a once-in-a-generation opportunity to fly a low-cost asteroid cratering experiment