



GPU Accelerated 3-D Modeling and Simulation of a Blended Kinetic Impact and Nuclear Subsurface Explosion

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Introduction

- Last minute scenarios with short warning time require a large amount of energy to be transferred to the NEO.
- Highest technological readiness for deflection options include kinetic impactors and explosives – highest energy density of available payloads.
- David Dearborn and others have shown that complete disruption of a body is not always an undesirable effect.
- Dispersion along the NEO orbit can substantially reduce the amount of mass remaining on impacting trajectories.

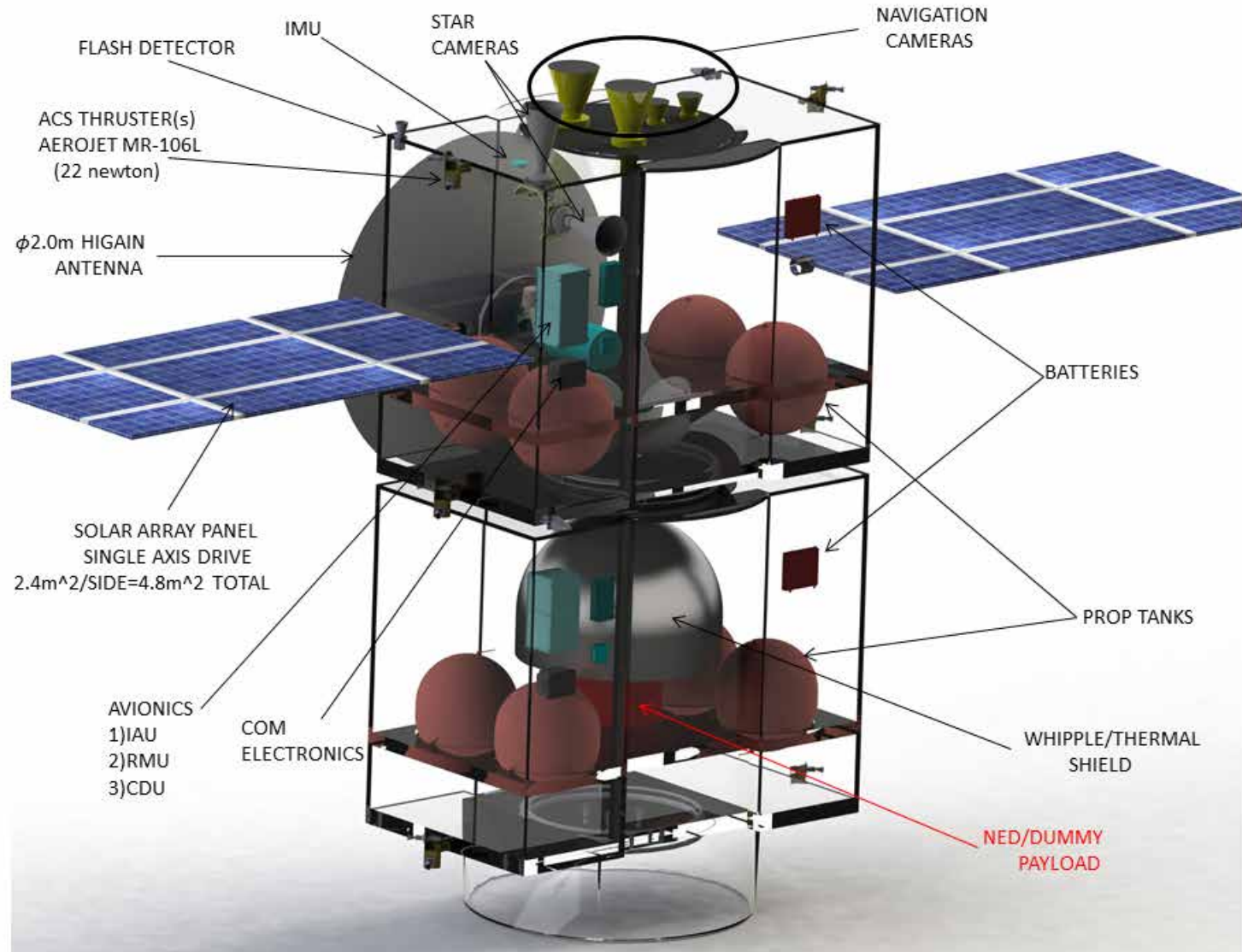
Introduction

- Energy coupling through subsurface explosions is substantially higher than other options, but this requires a rendezvous-type mission.
- Changing the arrival velocity can directly affect mission feasibility, particularly at the last minute.
- Hypervelocity Asteroid Intercept Vehicle (HAIV) attempts to work around this velocity limitation by blending the concepts of a kinetic impactor and a subsurface explosion.

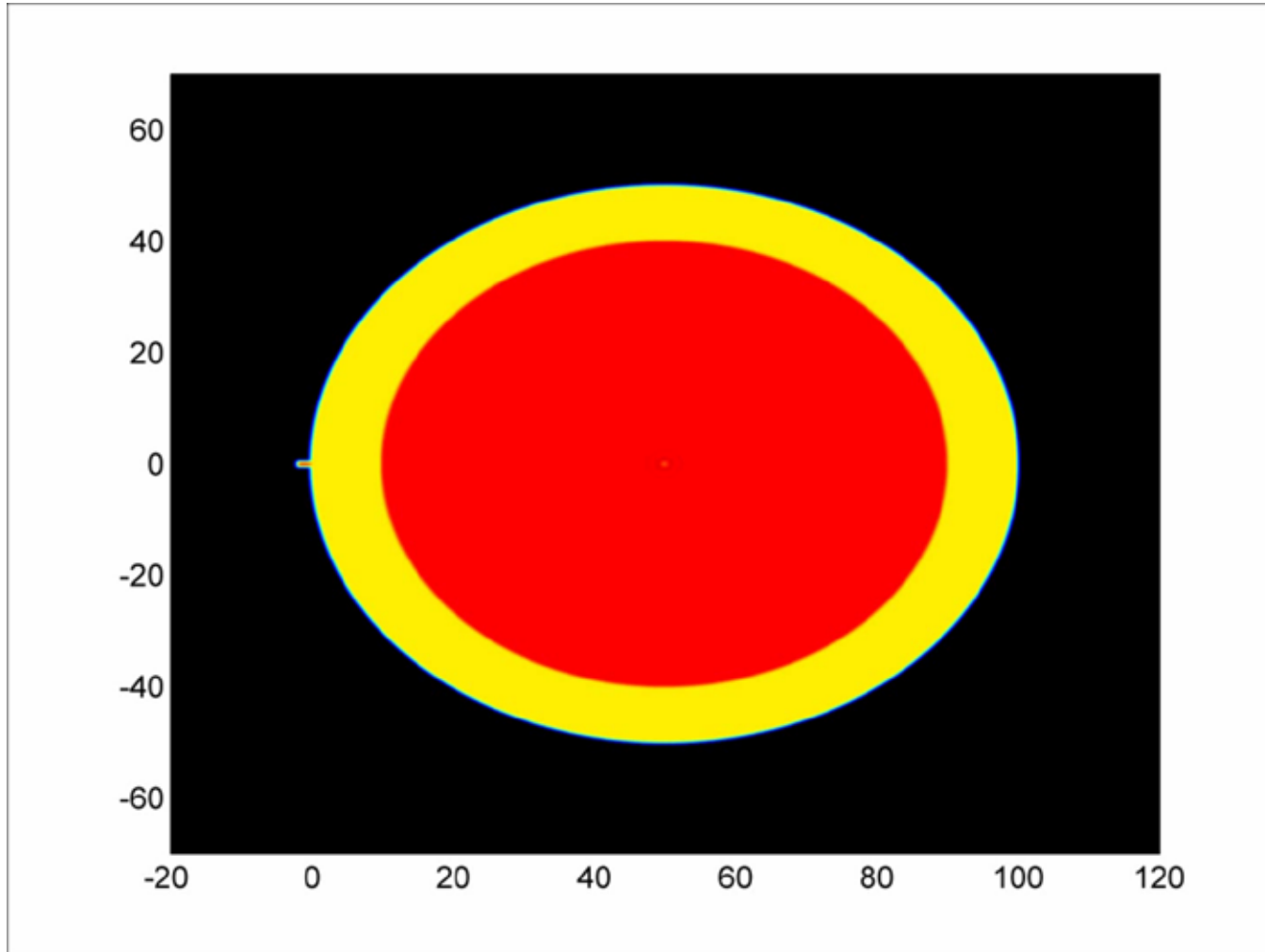
Introduction

- HAIV consists of two bodies. One impacts the target and forms a crater. The second detonates inside the crater, simulating a subsurface blast without the required velocity change.
- The present computational model simulates NEO fragmentation using subsurface, impacting, and standoff explosions.
- Focus on high-performance computing of these simulations allows us to integrate efficacy simulation into the mission design cycle.

Hypervelocity Asteroid Intercept Vehicle (HAIV) Concept



Introduction



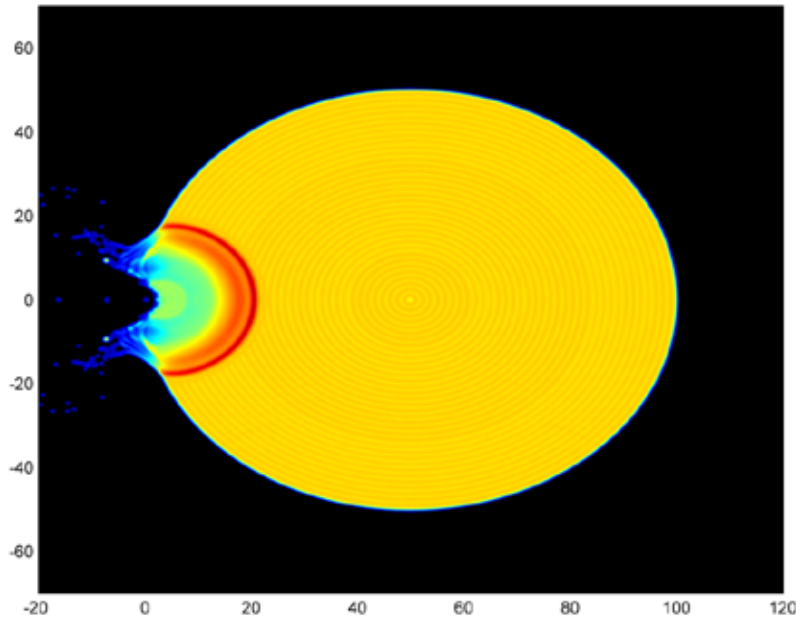
Spherical Target Model Description

- 100 meter diameter reference targets.
- Source energy equivalent to 100 kT.
- “Rubble-Pile” target with density of 1.91 g/cm^3 (no material strength model)
- Solid granite target with density of 2.6 g/cm^3 (linear elastic-plastic strength model)
- “Mixed” target with granite core and shell of rubble.
- Desired Resolution for largest cells: 0.1 m

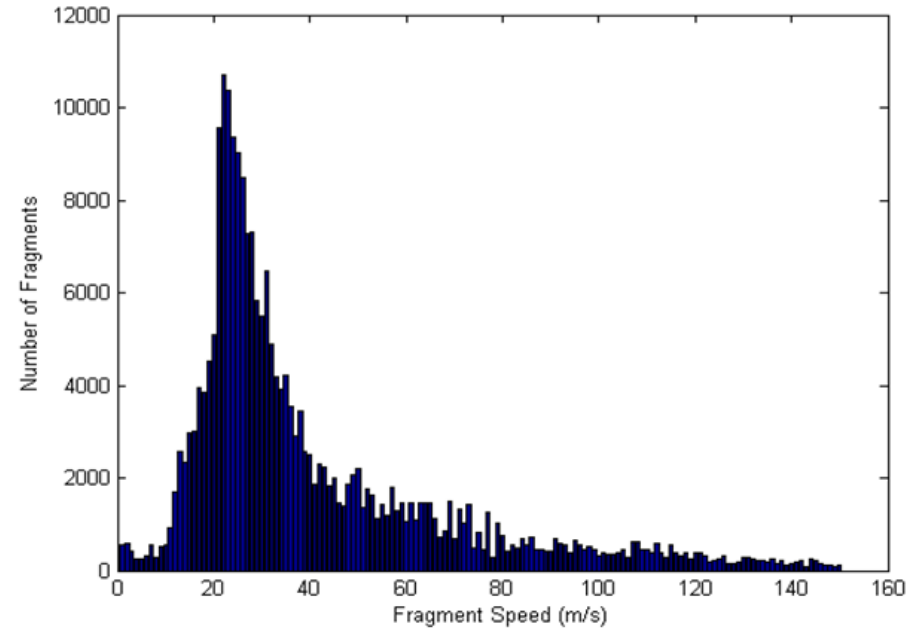
Energy Source Description

- 5 different disruption methods were tested on the two reference targets.
- A static, “buried” explosive at 5 meter depth.
- Static, surface blast with no relative velocity, explosion surrounded by aluminum projectile casing.
- Dynamic blast at surface with 6.1 km/s velocity.
- Two impactor case, penetrator followed by explosive.
- “Standoff” blast at 10 m height to ablate surface material.

Fragmentation Model Description

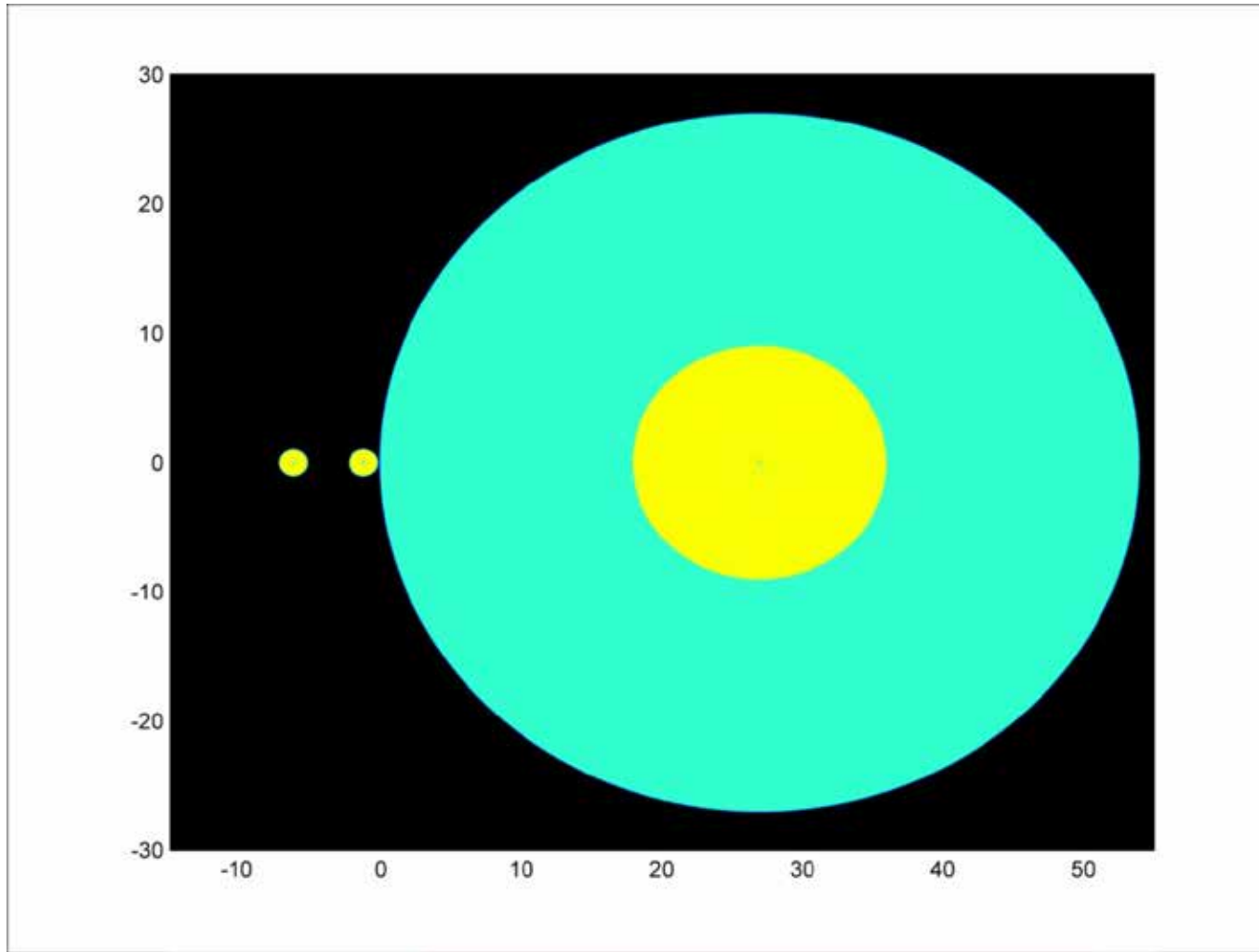


Subsurface Explosion in Solid Target

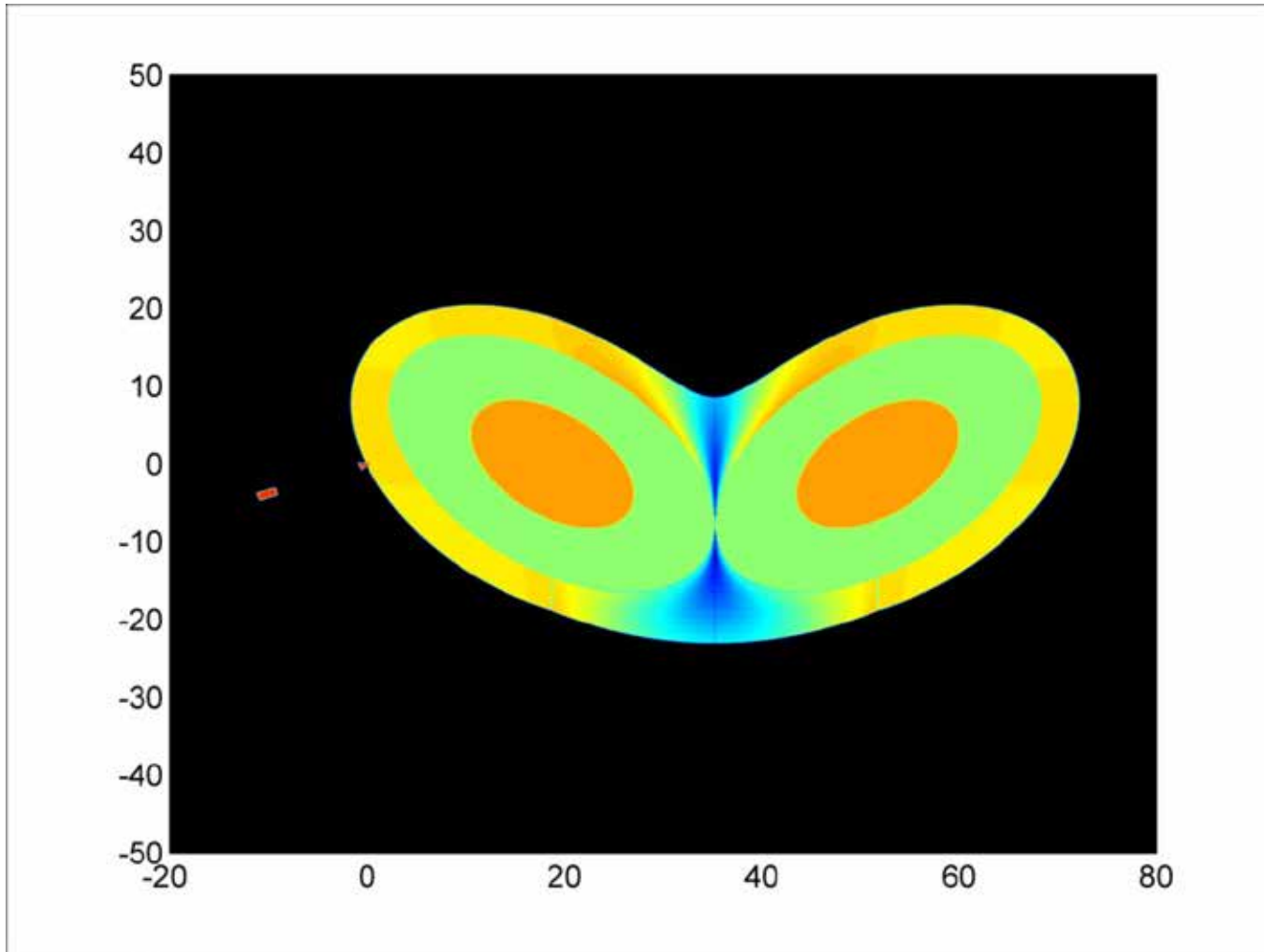


Resulting Fragment Velocities

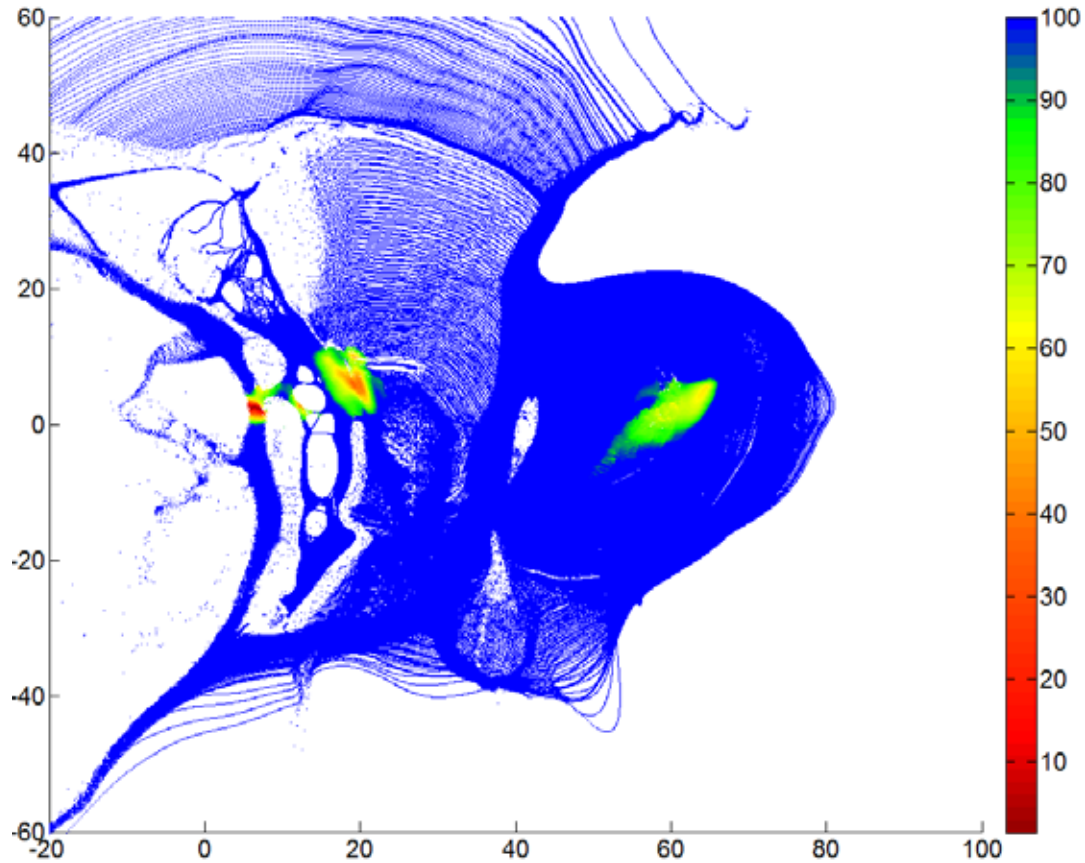
Fragmentation Model Description



Fragmentation Model Description

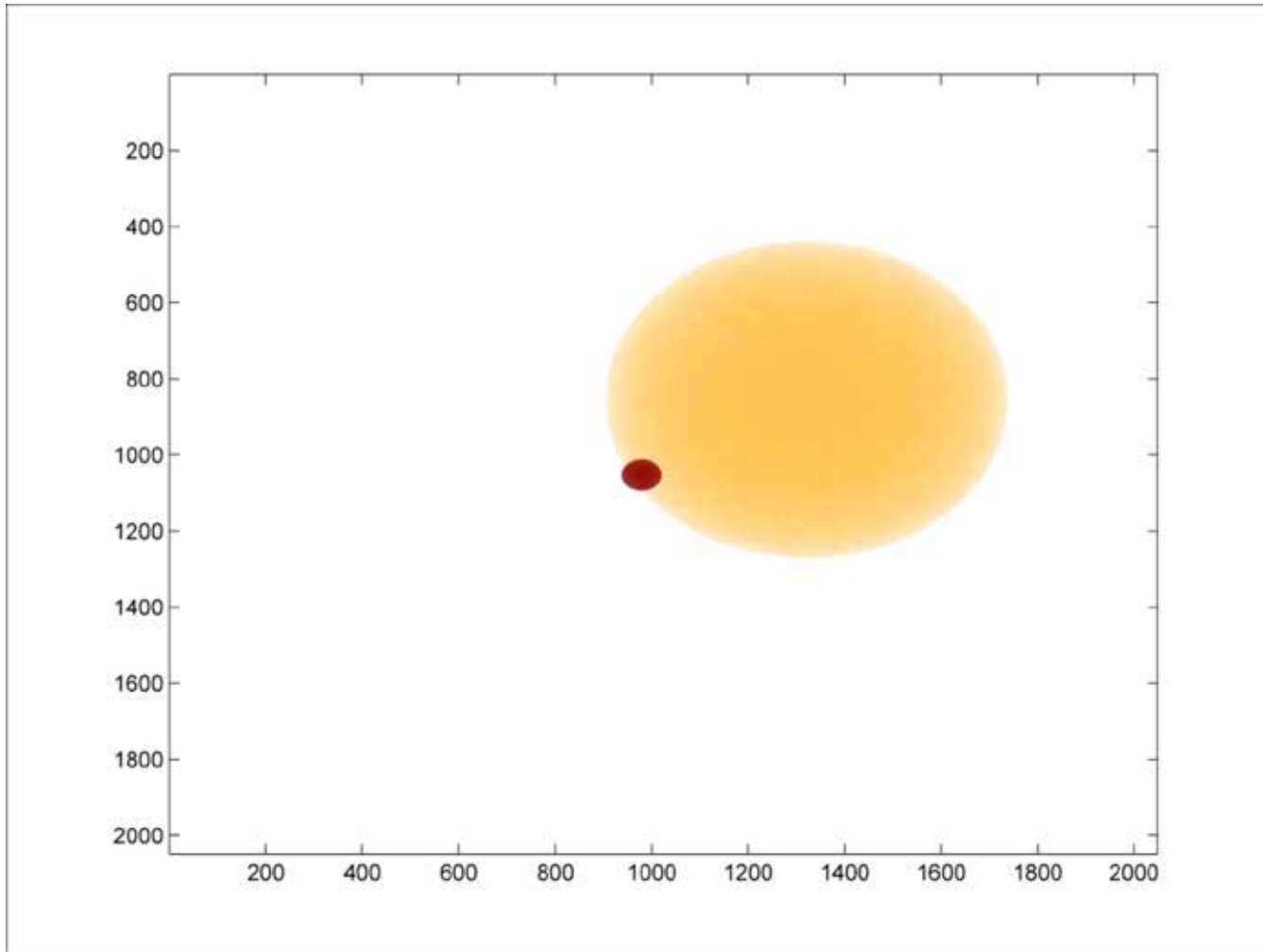


Fragmentation Model Description

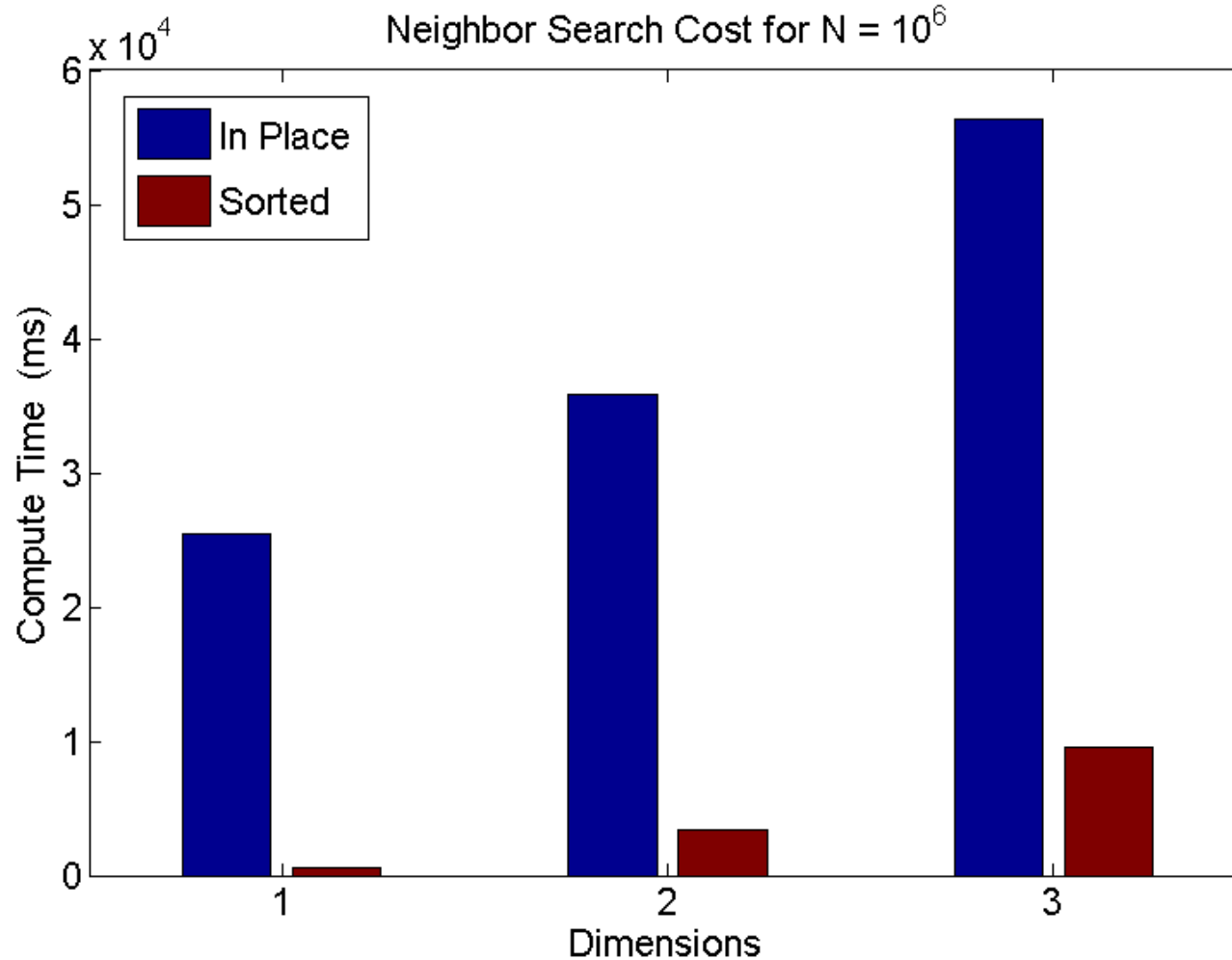


Slowest Moving Debris

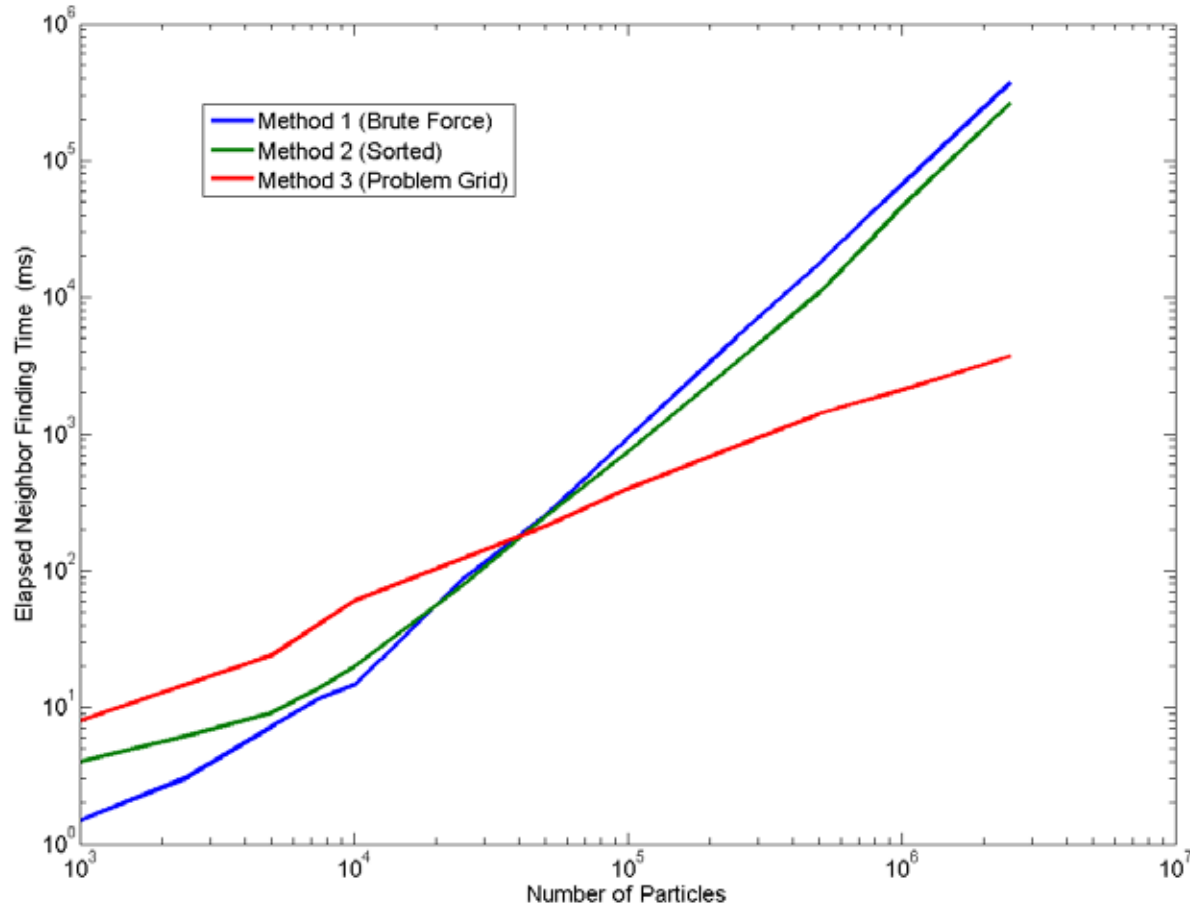
Fragmentation Model Description



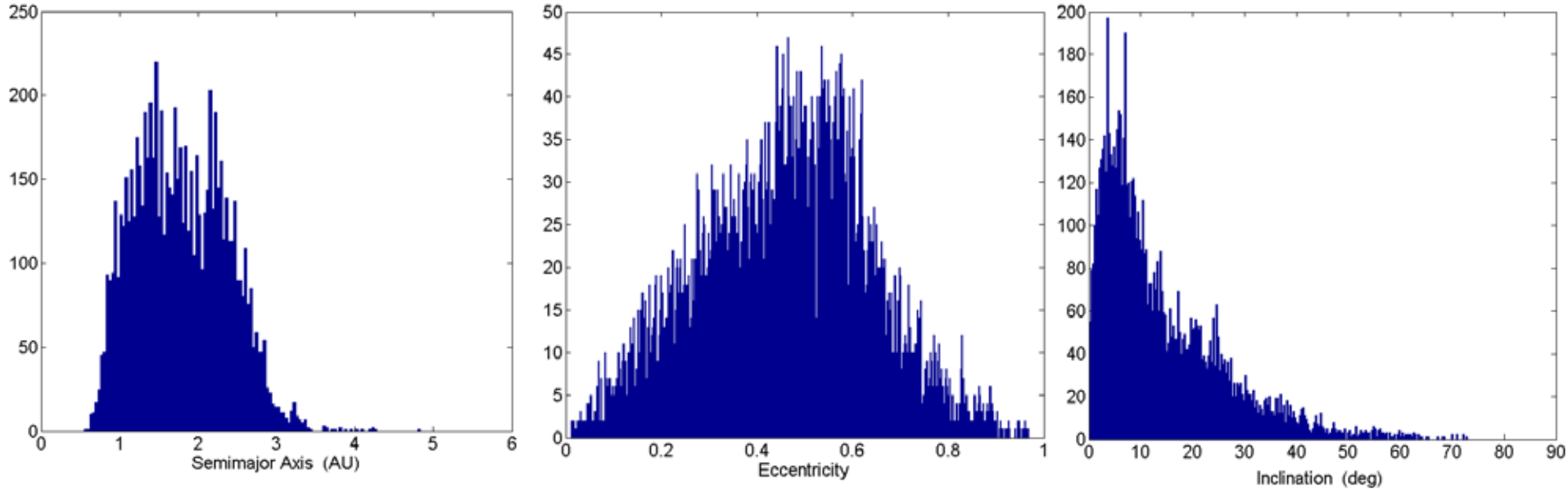
Neighbor Search



Neighbor Search (Cont)

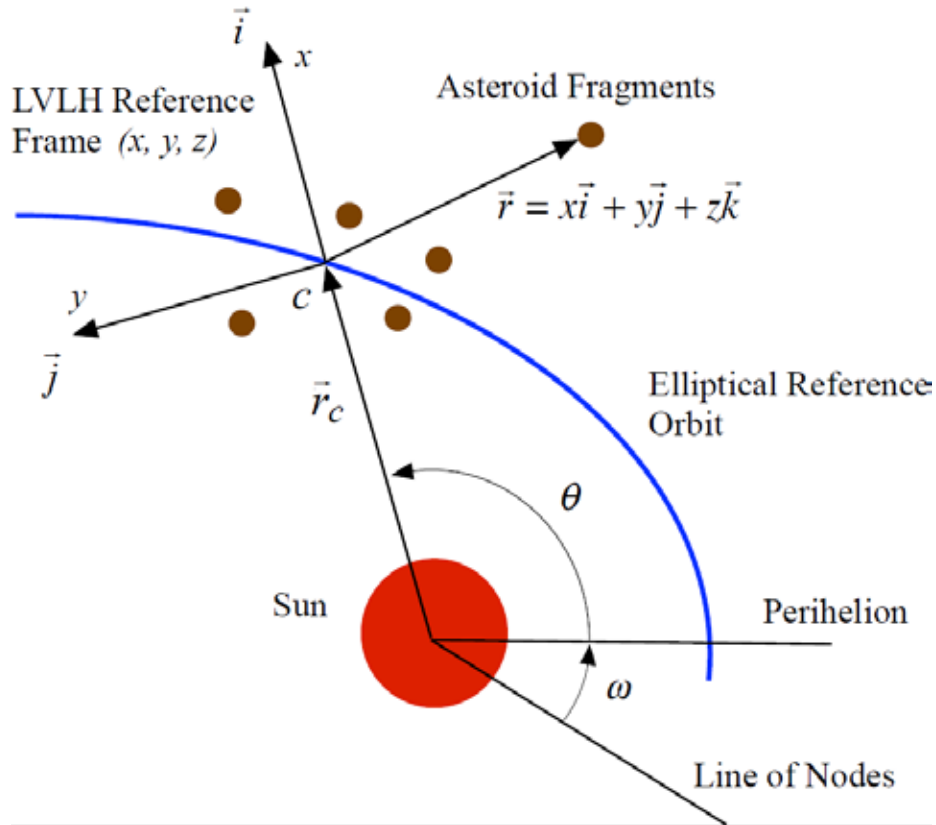


Virtual Impactor Orbits



Orbital Parameter Histograms of Observed Earth-Crossing Asteroids

Orbital Equations of Motion

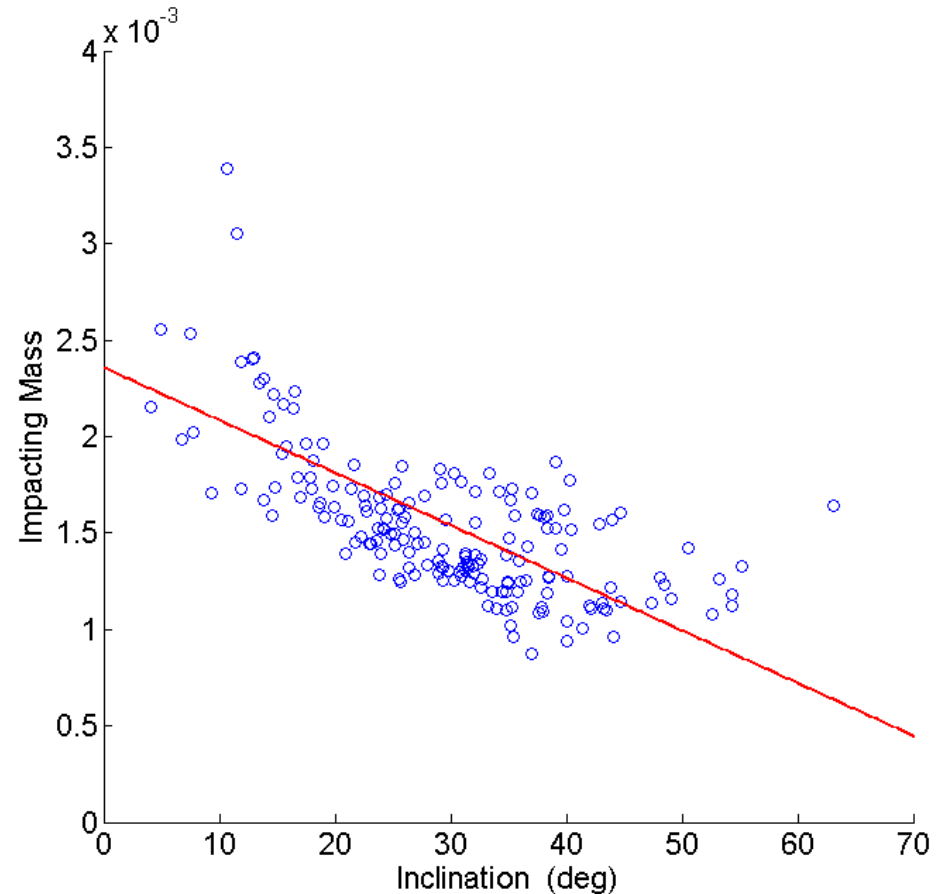


- Relative motion form of equations useful for mutual gravitational terms and collision modeling.
- Integrator uses LVLH coordinate system (shown).
- Simulated deflections occur along radial, transverse, and normal axes. In general, radial is found to be near optimal for 15 day lead time.

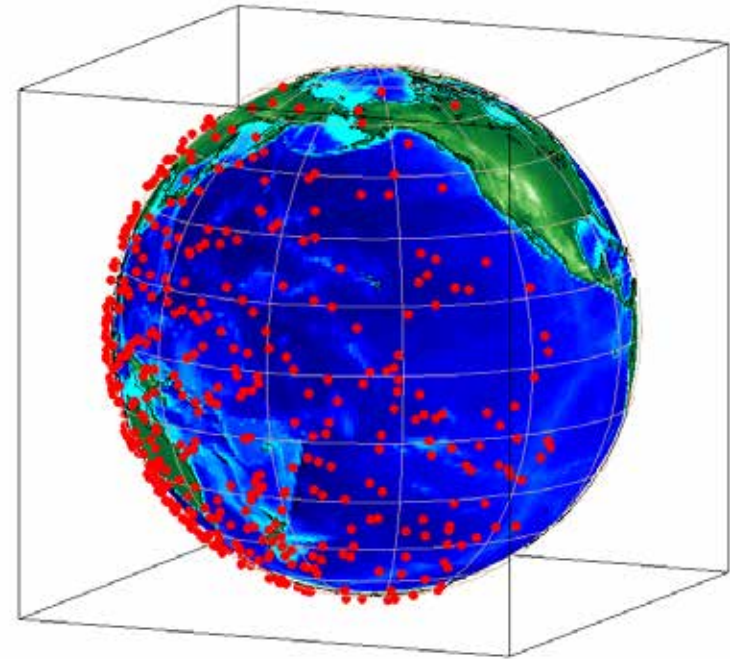
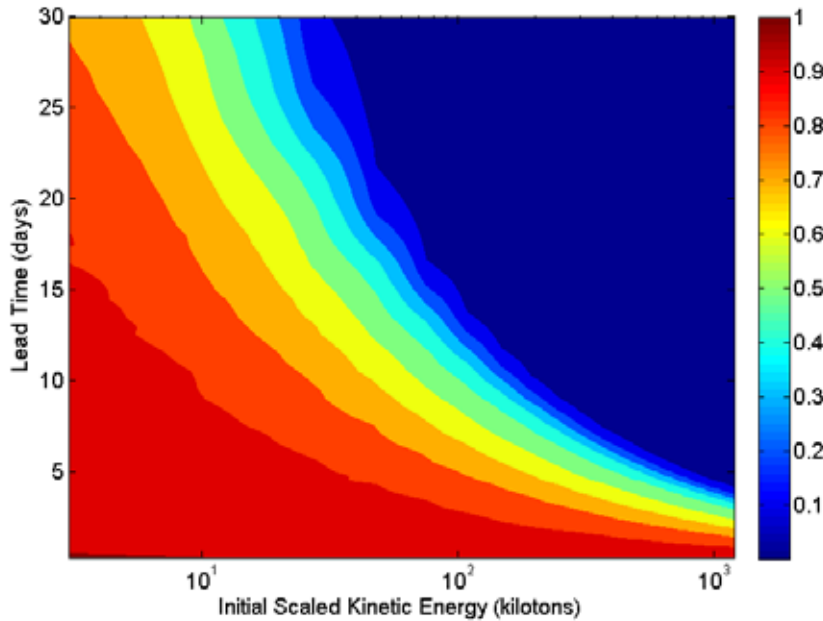
LVLH Coordinate System for Fragment Dispersion Model

Orbital Analysis Results

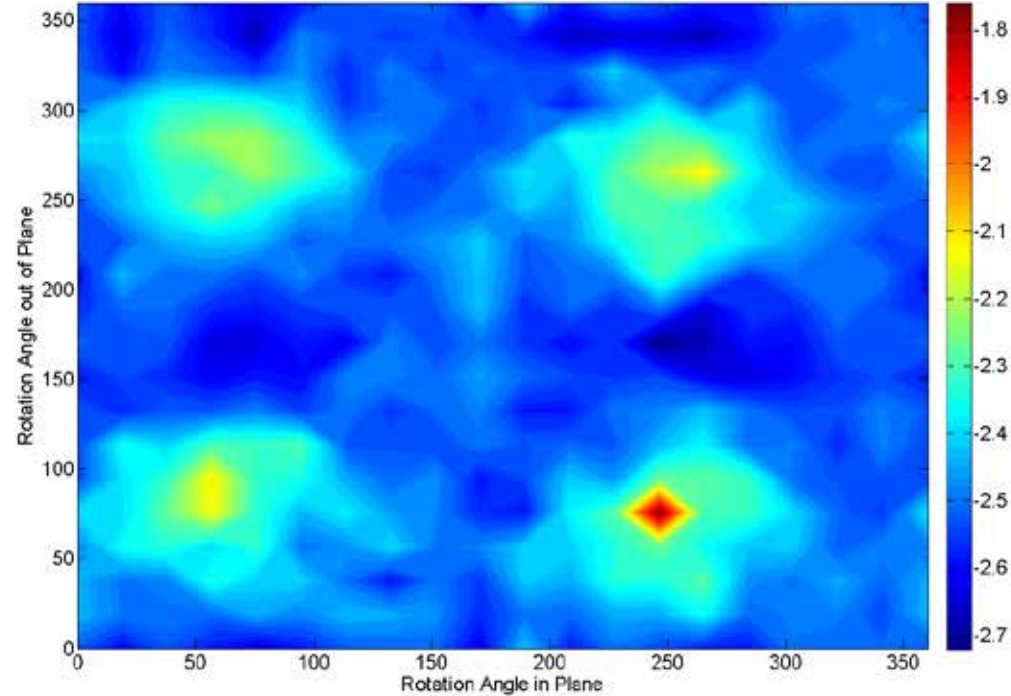
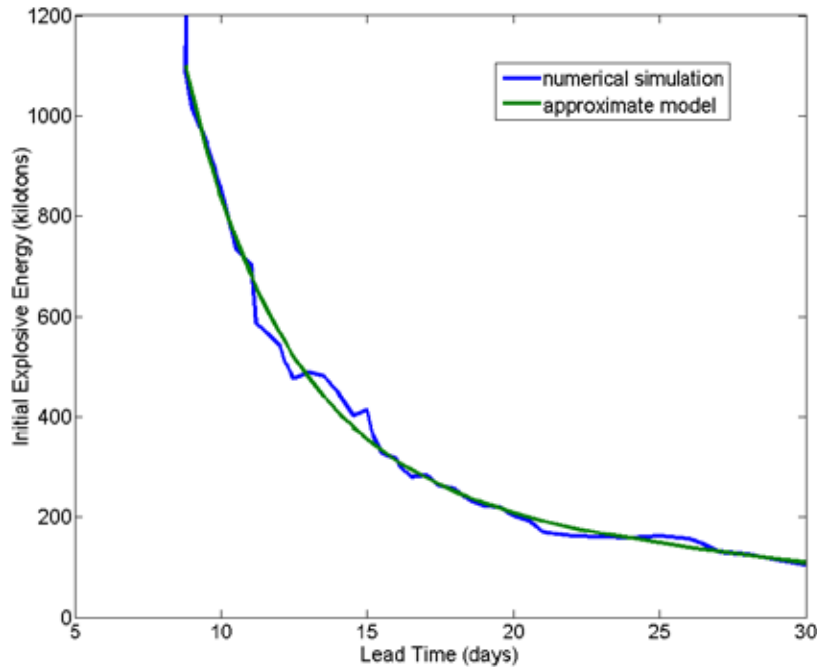
- No strong correlation was found between semimajor axis or eccentricity and impact probability for 15 day lead time.
- Deflection on higher inclination orbits were more effective.



Orbital Analysis Results

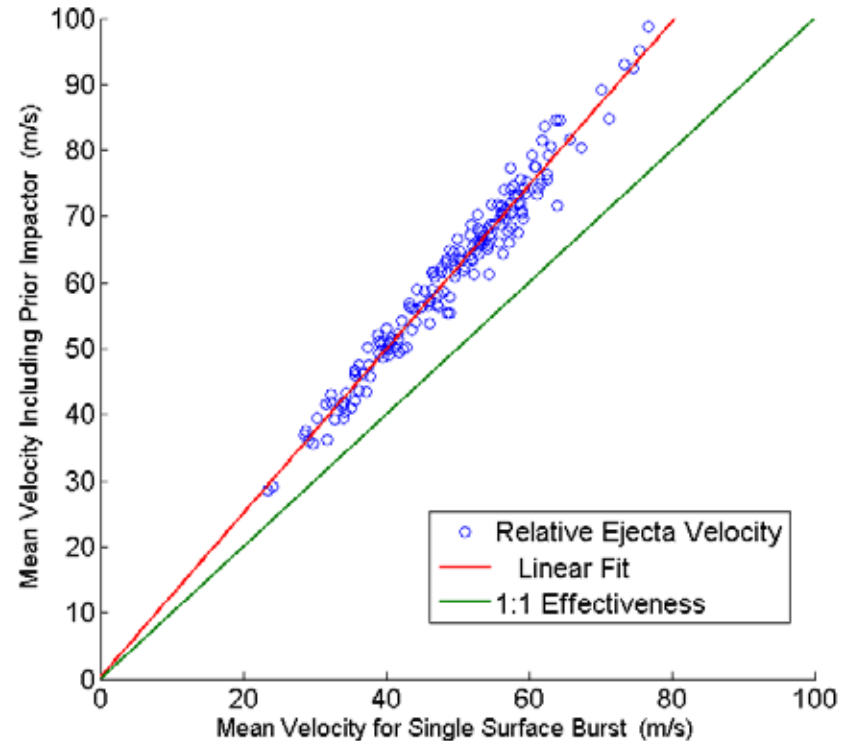


Orbital Analysis Results

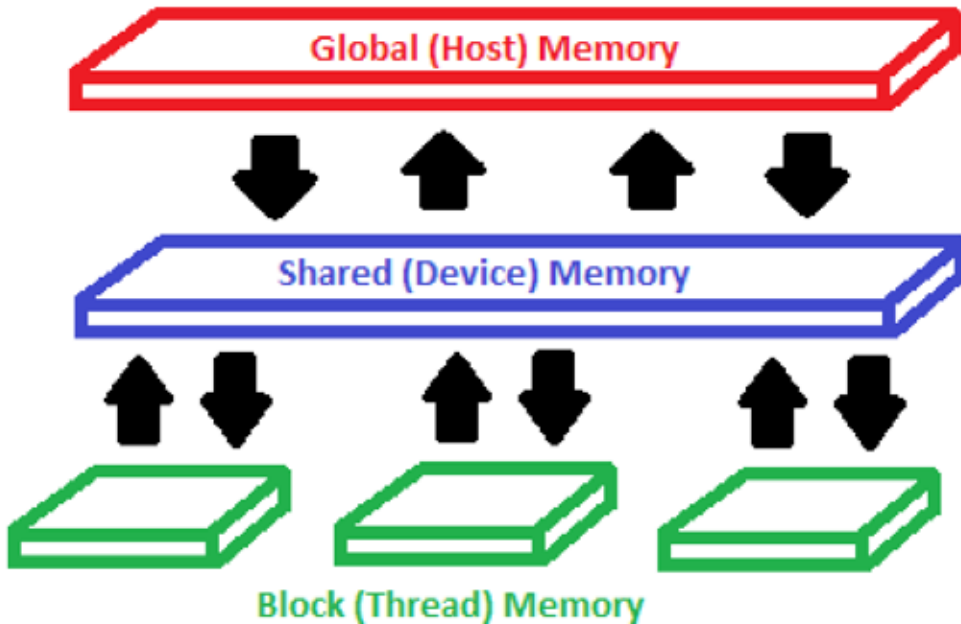


Orbital Analysis Results

- Two impactor approach is viable from an effectiveness standpoint.
- 25% better dispersion velocities on average than single surface blast.
- Major problems with GNC for this arrangement, particularly with uncertainty.



Computational Approach

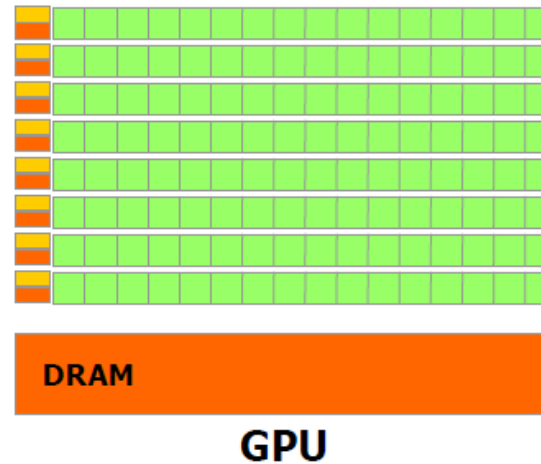
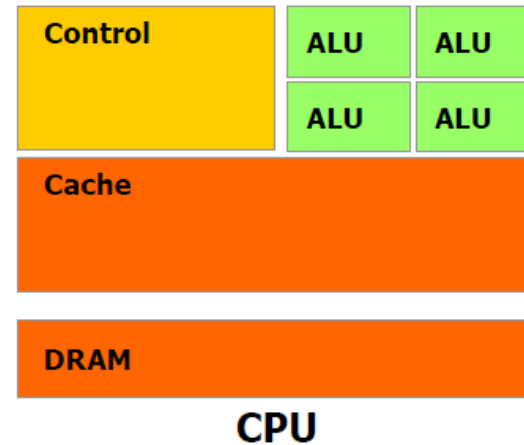


Qualitative GPU Memory Model

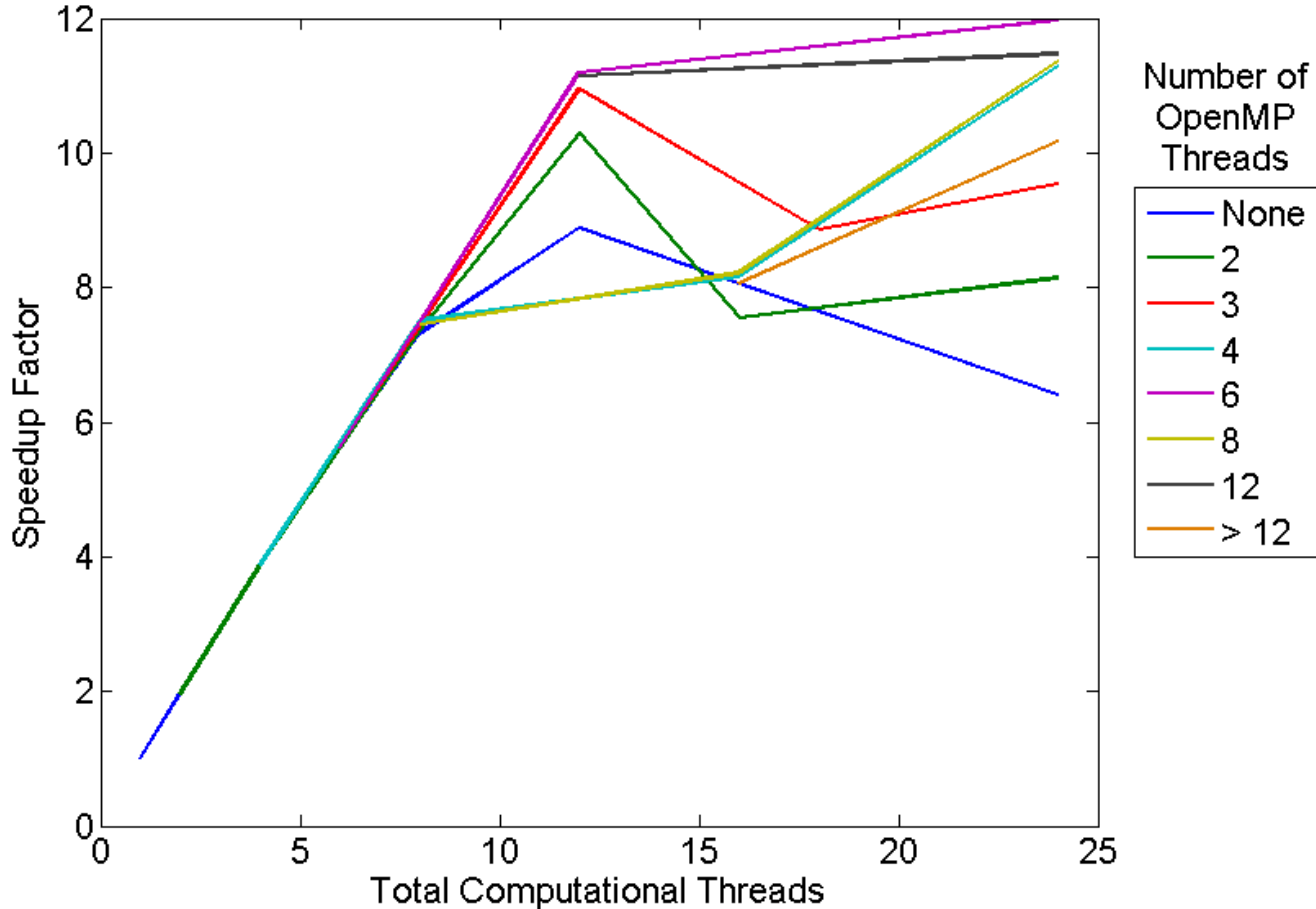
- Evaluating performance on single node of planned GPU computing cluster.
- 3 approaches: MPI, Hybrid with OpenMP, CUDA.
- Identify threading relationship yielding best speedup, and limitations of MPI/GPU implementation.
- Make use of memory footprint optimization to increase available resolution. Current limits ~6M part/node or ~18M part/node at 30% performance.

Computational Approach (Cont.)

- Huge difference, particularly in memory design.
- Core-level cache, common in CPU design, not present on the GPU.
- More than one thread per core in general, due to different floating point operators.
- Example: 4x improvement fusing access for cores rather than independent memory.

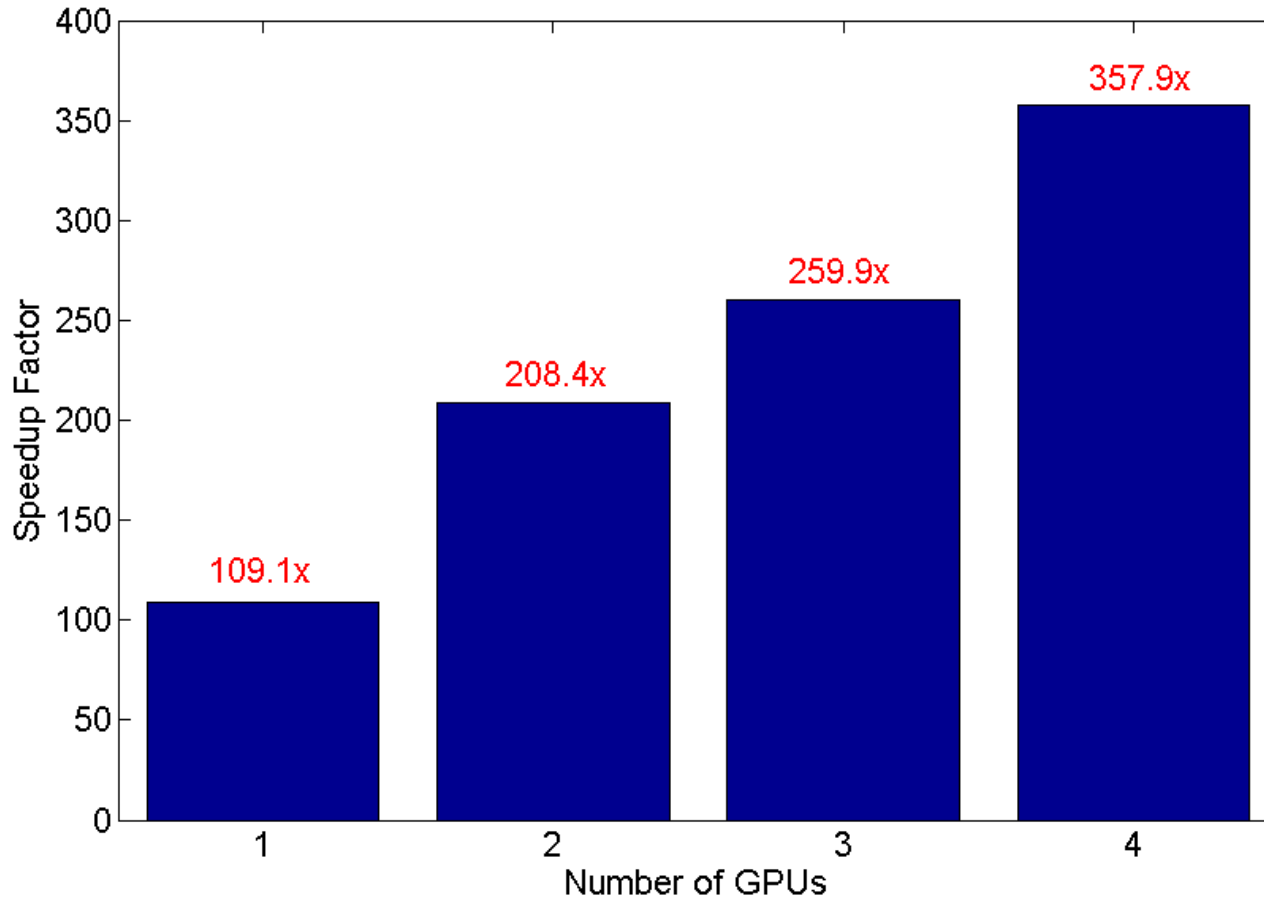


Results – CPU Performance



Speedup Factors for Several MPI and Hybrid Schemes

Results – Performance (Cont.)



CUDA Algorithm Speedup Factors (N=3.1M)

Conclusions

- An improved model has been developed and tested to simulate hypervelocity impact and explosive disruption.
- SPH presents a viable option for the simulations needed by ADRC. The present model agrees strongly with work done at LLNL.
- Statistical representation of observed NEO orbits indicates that a 100 m target could be safely disrupted on all orbits with 15 days of lead time.
- Cost effectiveness in relation to comparable top-of-the-line cluster architecture is 15x.
- Additional uses for GPU computing for simulations of interception and characterization have been identified.



Acknowledgements

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Questions?

- Kaplinger, B.D., Wie, B., and D. Dearborn, "Earth-Impact Modeling and Analysis of a Near-Earth Object Fragmented and Dispersed by Nuclear Subsurface Explosions," published in *The Journal of the Astronautical Sciences*.
- Kaplinger, B.D., Wie, B., and D. Dearborn, "Nuclear Fragmentation/Dispersion Modeling and Simulation of Hazardous Near-Earth Objects," published in *Acta Astronautica*.
- Pitz, A., Kaplinger, B., Vardaxis, G., Winkler, T., and B. Wie, "Conceptual Design of a Hypervelocity Asteroid Intercept Vehicle and Its Flight Validation Mission," to appear in *Acta Astronautica*.
- 19 Related Conference Papers.



Hardware Table

System	Machine 1	Machine 2	Machine 3	Machine 4	Machine 5
CPU	1x Core2 Q6600	1x Core2 Q6600	1x Xeon X5550	2x Xeon E5520	2x Xeon X5650
CPU Cores	4	4	4	8	12
CPU TPEAK	9.6 GFLOPs	9.6 GFLOPs	12.8 GFLOPs	21.36 GFLOPs	32.04 GFLOPs
GPU	1x 8800GTS	1x GTX470	1x GTX480	4x Tesla c1060	4x Tesla c2050
GPU Cores	112	448	480	960	1792
GPU TPEAK	84 GFLOPs	324 GFLOPs	385 GFLOPs	336 GFLOPs	2060 GFLOPs
CUDA CC	CC 1.0	CC 2.0	CC 2.0	CC 1.3	CC 2.0