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SOLAR-SAILING TRAJECTORY DESIGN FOR CLOSE-UP NEA OBSERVATIONS MISSION

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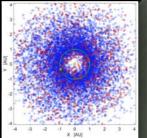
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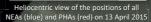
Why a multiple NEA rendezvous mission through solar sailing?

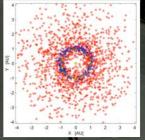
A good knowledge of chemical and physical properties of NEAs is needed in order to plan any mitigation mission. In order to improve our knowledge about NEAs, a rendezvous with these objects is better than ground-based observations, although more expensive. Moreover, the number of possible NEAs to visit is very large.

Multiple NEA rendezvous missions can be a way to visit as many objects as possible within one single mission, thereby reducing the cost for a single rendezvous.

Because of the high Δv required, solar sailing can be a good way to perform this kind of mission.







in blue and PHAs in red) on 13 April 2015

 $a \in [a_{min}, a_{max}]$

 $e \in [e_{min}, e_{max}]$ $e_1 - \varpi_2| \le \delta \varpi_{max}$

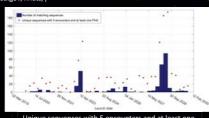
 $|\hat{h}_1 \cdot \hat{h}_2| \le cos(\vartheta_{max})$

- A tree search algorithm has been developed in order to find as many sequences of NEAs as possible within the mission requirements underlined in the DLR/ESA Gossamer roadmap [1].
- An automatic tool has been developed in order to optimize the sequences chosen from the previous output.
- A solar sail with a lower area-to-mass ratio, with respect to Ref. [1], has been taken into consideration, so that more payload can be carried on or a smaller sail can be used, raising the TRL.

Sequence search

Starting from a database made of NHATS[†] asteroids and PHAs only and pruning the temporary database by means of astrodynamical considerations, a systematic search of sequences has been carried out with launch dates from 28 Nov 2019 for 10 years with 3 months

Near-Earth Object Human Space Flight Accessible Targets Study (http://neo.jpl.nasa.gov/nhats/)



PHA found by the sequence finder algorithm

Complete database (NHATS + PHAs) Local pruning on Keplerian elements

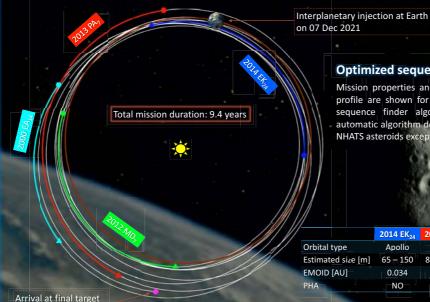
Semi-analytical approximated trajectory (shape-based method) from Earth to all available NEAs

Local pruning on Keplerian elements

Semi-analytical approximated trajectory (shape-based method) to all available NEAs

Add stay time at the object

Mission time > 10 years?



Optimized sequence test case

Mission properties and an overview of the overall mission profile are shown for one of the sequence found by the sequence finder algorithm and optimiz through the automatic algorithm developed. Note that he objects are NHATS asteroids except 2000 EA₁₄, which ssified as PHA.

2014 EK₂₄ 2013 PA₇ 2012 MD₇ 2000 EA₁₄ 2008 DB Orbital type ollogA Aten Amor ollogA Estimated size [m] 65 - 150 85 - 190 40 - 95 170 - 37020 - 50EMOID [AU] 0.034 0.091 0.018 0.043 0.002 NO NO NO YES-NO

on 26 Apr 2031

[1] Dachwald, B., Boehnhardt, H., Broj, U., Geppert, U. R. M. E., Grundmann, J.-T., Seboldt, W., Seefeldt, P., Spietz, P., Johnson, L., Kührt, E., Mottola, S., Macdonald, M., McInnes, C. R., Vasile, M. and Reinhard, R., "Gossamer Roadmap Technology Reference Study for a Multiple NEO Rendezvous Mission," Advances in Solar Sailing, edited by M. Macdonald, Springer Praxis Books, Springer Berlin Heidelberg, 2014, pp. 211-226.

