

A visualization of space debris in Earth's orbit. A central globe of Earth is surrounded by a dense cloud of small grey dots representing debris. Several larger, faint elliptical orbits are visible, showing the paths of satellites and other objects in space.

6th European Conference on Space Debris

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- **6th European Conference on Space Debris:**
 - held at ESA/ESOC, Darmstadt/Germany, 22-25 April 2013
 - organized by the ESA Space Debris Office
 - co-sponsored by ASI, CNES, DLR, UKSA, COSPAR, and IAA
- **participation statistics:**
 - 355 participants from 26 countries (record participation)
 - 300 abstracts received
 - 115 oral presentations in 13 technical sessions
 - proceeding to be distributed as „ESA-SP 723“ before end Sep. 2013
- **concluding press conference:**
 - panel members: C. Bonnal (CNES & IAA), H. Lewis (UKSA), C. Portelli (ASI), T. Schildknecht (COSPAR), M. Metz (DLR), H. Klinkrad (ESA & IAA)
 - very good media coverage throughout the Conference

■ highlights & core findings [1]:

- **measurements & surveillance:** progress on radar observation technologies was reported, particularly from Japan and China; a strong interest, was noted for optical systems, in the context of tracking and surveillance applications, also in low orbits, with active illumination of targets by laser beams
- **modeling:** consolidated information was provided by studies of IADC, national entities, and academia on the long-term evolution of the orbital debris environment and the suspected start of a run-away situation (known as the “Kessler Syndrome”) at 700km to 1000km altitude, leading to collisional cascading within a few decades; the reported analyses suggest that the only effective remediation action would be active debris removal at rates of 5 to 10 large objects per year
- **mass removal:** concepts were outlined that employed a whole range of different physical principles to initiate an immediate, controlled, or to a delayed re-entry; mass removal concepts comprised net or harpoon-based capture & tug servicers, attached propulsion units, drag augmentation, conductive tethers, or contact-less momentum exchange through plume interaction with chemical or ion thrusters

■ highlights & core findings [2]:

- **on-orbit risk reduction:** debris avoidance techniques and the required accuracy and information content of underlying data were highlighted; CNES and ESA explained their use of JSpOC (US Joint Space Operations Center) conjunction messages to assess identified close conjunctions, based on derived risk levels; for the protection against sub-catalog object sizes deployable shield structures were introduced; HVI studies also indicated that impact ejecta will further deteriorate the small-particle environment, and that impact-induced plasmas may cause damage to operational spacecraft.
- **re-entry risk assessment:** several authors reported on decay and re-entry prediction methodologies, on methods for on-ground casualty estimations, and on the determination of casualty cross-section and high-resolution population density data to support such risk assessments; the concept and application of a re-entry break-up recorder was outlined to improve our understanding of physical processes during the re-entry and break-up phase
- **policies & standardization:** economic aspects of controlling the space debris environment were addressed; also legal and policy aspects of SSA systems and of debris remediation were highlighted